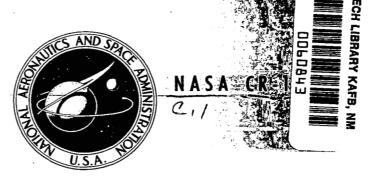
NASA CONTRACTOR REPORT





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COMBUSTION PROCESS OF IMPINGING HYPERGOLIC PROPELLANTS

by L. B. Zung and J. R. White

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FOREWORD

The research described in this report was conducted by the Dynamics Science Division of Marshall Industries under NASA contract NAS 3-12031. Mr. Richard J. Priem of the Lewis Research Center Chemical Rocket Division was the NASA Project Manager. The report was originally issued as Dynamic Science report SN-145-F.

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SUMMARY

An experimental study was conducted to study the spray formed by impinging streams of liquid hydrazine and nitrogen tetroxide. The experimental technique was based on photographic observation of the resulting liquid spray. Single exposure color photographs and high speed motion pictures with simultaneous chamber pressure traces recorded on the film were used.

Stream mixing and separation phenomena were determined based on the photographic observations. At low chamber pressures, the boiling of nitrogen tetroxide was found to play a significant role. With nitrogen tetroxide above the boiling point the resulting spray pattern showed that most of the liquid hydrazine was confined within the fuel side and thus resulted in very poor mixing. At high chamber pressure, separated flow was also observed. The resulting spray definitely showed a color variation with colorless hydrazine mainly on the fuel side of the spray and liquid nitrogen tetroxide on the oxide side. Good liquid spray mixing was observed at lower pressures. Propellant temperatures and jet diameter to velocity ratio appeared to be unimportant. Propellant additive also did not influence mixing at 15 Psia.

The occurrence of combustor pressure popping (sudden rise in chamber pressure which accompanied rapid burning of the liquid) was also found to depend on chamber pressures. Using .055" and .060" orifices, combustor pressure popping was very significant for chamber pressures below approximately 190 Psia and popping was rarely observed for chamber pressures above 190 Psia. Popping frequency and the average overpressure attained lower values at higher chamber pressures. Injector orifice played a significant role. Using .027" orifice, injector poppings were not observed and for .040" orifice, poppings were only observed infrequently under atmospheric pressures and were absent at high chamber pressures. To summarize, injector poppings occurred more frequently using larger orifice diameters at lower chamber pressures and less frequently using smaller orifice diameters at higher chamber pressures.

INTRODUCTION

Cold flow tests have been conducted to understand and to improve injector design technology. After very extensive studies of the resulting sprays formed by impinging chemically nonreactive streams, Rupe (Ref. 1) formulated an optimum mixing criteria. With impinging streams of real propellants, significantly different phenomena are observed, namely separation. As first reported by Elverum and Standhammer (Ref. 2) reactions of hypergolic propellants were rapid enough to prevent normal liquid phase mixing. These separation effects were particularly evident with hydrazine/nitrogen tetroxide systems. Continued experimental studies by Johnson (Ref. 3) and Evans, Stanford, and Riebling (Ref. 4) have shown reduced rocket engine performance with stream separation.

Photographic observations of stream separation phenomena were first conducted at NASA Lewis Research Center by Burrows (Ref. 5). Subsequent photographic studies of the resulting sprays were carried out at Dynamic Science (Refs. 6 and 7) to arrive at more quantitative data of the stream impingement phenomena. A slot injector was used with the resulting spray flowing over a lucite plate. Additional photographic observations have also been carried out at Rocketdyne (Ref. 8) and Aerojet (Ref. 9). An alternate method of using on-line mass spectrometer to study stream separation is being conducted by Houseman (Ref. 10).

Attempts to correlate and to predict stream separation phenomena have also been carried out. Separation criteria have been defined in terms of chemical reaction time and interface propellant residence time (Ref. 6). Kushida and Houseman (Ref. 11) proposed two complementary theoretical models; one based on the attainment of bubble point temperature, and a second model more applicable for higher pressures emphasized the gas phase reactions.

In addition to stream mixing and separation phenomena, violent explosions of liquid propellant sprays, commonly known as poppings, have also been observed. The objective of the present work was to determine the governing factors affecting stream mixing/separation phenomena and the occurrence of popping and to define engine operating regions where these different combustion processes are most likely to occur. These results will be most helpful for design engineers.

EXPERIMENTAL APPARATUS

Combustion Chamber

Impingement experiments were conducted using a single element, like on unlike, combustion chamber as shown in Figure 1. The injector consisted of a pair of stainless steel tubings press fitted into an aluminum injector plate, (Fig. 2). The tubings protruded 1/4 inch from the plate. Orifice alignments were properly refined under cold flow conditions by manually adjusting the protruding tubes. The impingement angle was fixed at 60° . The distance between the center of the two orifices were four orifice diameters apart. The ratio of the orifice length to that of orifice diameter (L/D) was equal to 100. Four different injector orifices, with orifice diameter respectively equal to .027", .040", .055", and .060" were used throughout the experiment.

The injector plate was fitted over a test chamber consisting of a rectangular aluminum block with a cylindrical hole 2" by 4 3/8" along the center of the block. Two viewing windows, with window openings equaled to 1.7" diameter were used for lighting and photographic purposes. Quartz windows were used because they were more compatible with the propellants as well as better light transmission in the ultra-violet range: Copper nozzles were attached to the bottom end of the chamber. Six different copper nozzles, with nozzle diameters equal to 1.5", .600", .400", .283", 179", and .128" were used to vary the chamber pressures.

Propellant Feed System

A schematic of the nitrogen tetroxide/hydrazine flow system is shown in Figure 3. The flow systems were fabricated with 304 stainless steel lines. The propellant tanks were designed for a working pressure of 1000 Psia. The tank capacities were approximately 1/3 gallon. Pressurization of the tank was controlled by a regulator and a release valve which was set at 1000 Psia. Two solenoid valves were used to vary the dome pressure of the regulator. Propellant flow rates were controlled by the propellant tank pressure and also by variable area cavitating venturi valves. The flow controllers were micrometer head type. This device provided easy and precise control of the propellant flows and allowed for more efficient testing with the ability to match desired stream momentum ratios. Heat exchangers were provided for each of the propellant tanks, and coaxial lines were used for the entire propellant supply system. Propellant temperature was controlled by flowing water at elevated temperatures or ethylene glycol-water mixtures at low temperatures through the heat exchangers and the propellant coaxial feedlines. Two 500 watt heaters were used to raise the water temperatures. A refrigerated coil bath conditioner, with a refrigeration compressor unit was used to condition the ethylene glycol-water mixture temperatures. Propellant temperature variations between 40°F and 140°F could be easily attained using this system. Prior to each test, the propellants were first led into a by-pass tank until the desired propellant flow rate and temperature were achieved before being introduced into the combustion chamber.

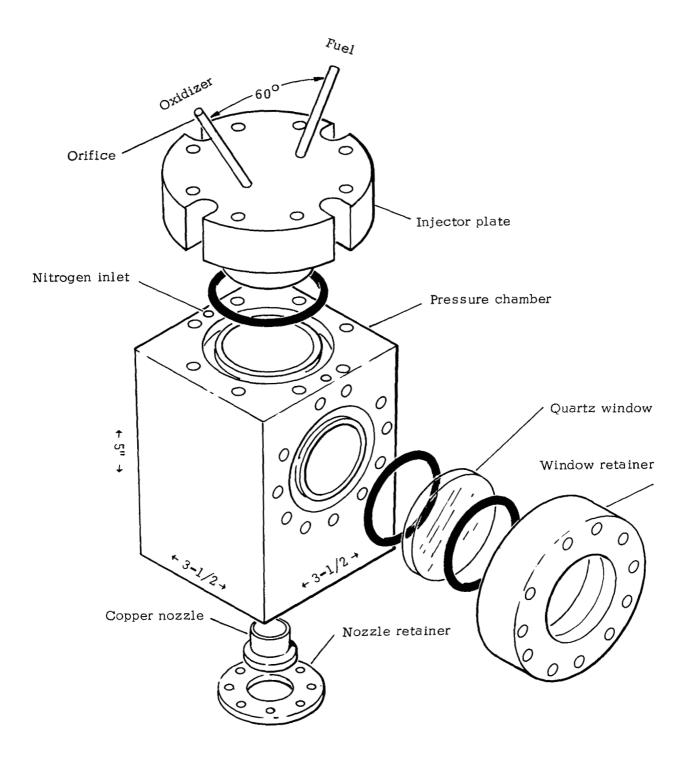
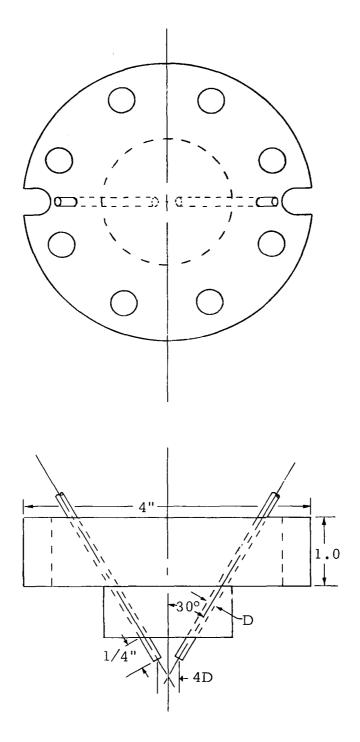


Figure 1. High Pressure Impingement Test Chamber.



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Figure 2. Injector Plate for Impingement Studies.

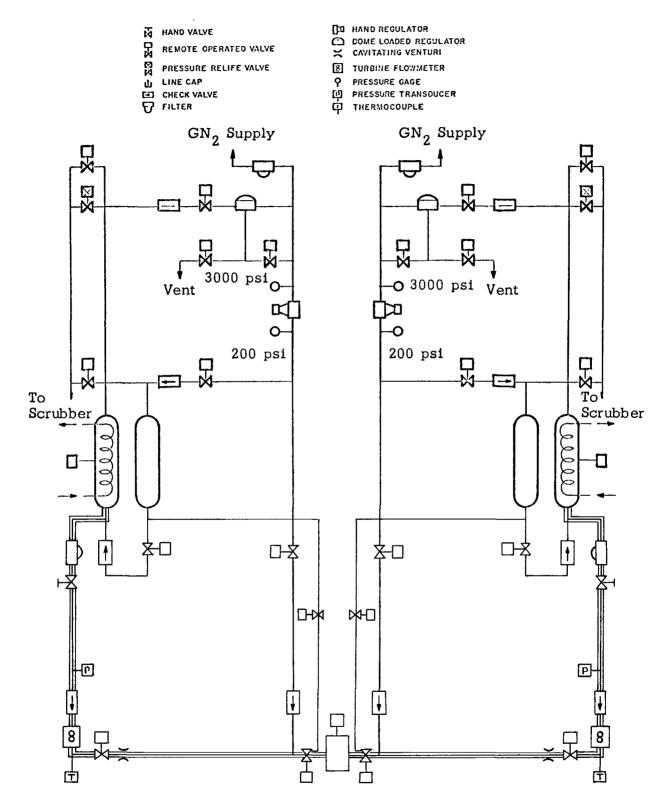


Figure 3. N_2O_4/N_2H_4 Flow System.

Recording Instrumentation

Hydrazine and nitrogen tetroxide flow rates were measured with turbine type flow meters. I/C thermocouples were used to record the propellant temperatures in the tanks; immediately downstream of the flow meters, and at the injectors. Propellant tank pressures were monitored by strain gauge transducers. Chamber pressures were measured with a quartz pressure transducer and a strain gauge transducer.

The test control console housed the propellant control console, a test sequence unit, amplifiers and various recording instrumentation and related components. The signals of the flow meters, thermocouples at the injectors, and pressure transducers of the combustion chamber were fed into amplifiers to a recording oscillograph. The flow rates were determined from the meter frequencies output and the propellant densities. An oscilloscope was also used in certain tests to record the pressure signal of the quartz transducer. Propellant tank pressures were recorded on a two channel strip recorder. Propellant temperatures in the tank and at the flow meters were recorded on a multipoint recorder. The entire experiment was operated remotely through the test control console.

Photography

The experimental technique was based on photographic observeration of the impingement region. Both single exposure photograph and high speed motion pictures were used. A microflash system was used to provide the proper lighting for single exposure photography. Flash duration was approximately 1.0 microsecond with peak light intensity of thirty million beam candlepower. This high intensity completely masked out the flame light, and the microsecond flash duration stopped the stream and droplets motion, thus allowing clear observation of liquid hydrazine and liquid nitrogen tetroxide. All of the single exposure photographs were recorded on color films.

The impingement phenomena were also recorded on high speed motion pictures. A rotating prism camera, with framing speed set at 3000,4000, and mostly at 5000 frames per second was used. In addition, a shutter with an exposure ratio of 1/100 was also employed so that a two microsecond exposure time was obtained with a framing speed of 5000 per second. To observe injector poppings, it was most desirable to correlate the chamber pressures with that of the explosion phenomena. To best achieve this correlation, the output of the quartz transducer was fed into an oscilloscope and the film was back-exposed with the oscilloscope trace. Thus the high speed motion pictures simultaneously recorded the injector poppings and the corresponding pressure spikes. Figure 4 shows the experimental setup where a second lens was attached to the backside of the movie camera and was used to focus the chamber pressure traces on the oscilloscope.

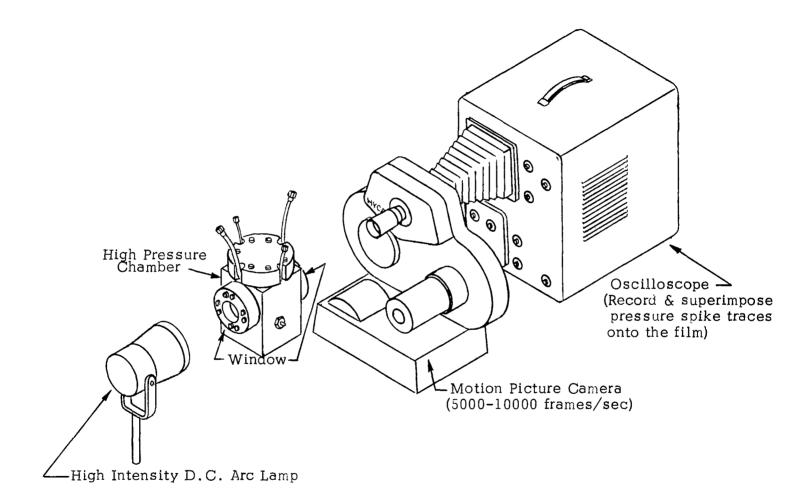


Figure 4. Experimental Setup for Studying Injection Mixing Explosions.

IMPINGEMENT EXPERIMENTS UNDER ATMOSPHERIC PRESSURE

Tests were conducted by impinging liquid streams of hydrazine and nitrogen tetroxide either in the open atmosphere or within the combustion chamber using a 1 1/2 inch diameter nozzle. The latter technique was desirable to protect the photographic instrumentation when closer observations were required. Only single exposure photography was employed during this phase of study. Results were analyzed based on the photographic observations of the spray by impinging streams of hydrazine and nitrogen tetroxide. The test parameters for this phase of study were as follows:

Chamber Pressure : one atmosphere

Propellant Temperature

Hydrazine : $45^{\circ}F - 105^{\circ}F$ Nitrogen Tetroxide : $42^{\circ}F - 80^{\circ}F$

Propellant Velocity

Hydrazine : 18 ft/sec - 94 ft/sec Nitrogen Tetroxide : 16 ft/sec - 72 ft/sec

Orifice Diameters (inch) : .027, .040, .055, .060.

Two distinctly different impingement phenomena were observed under atmospheric pressure tests. For nitrogen tetroxide temperature above the boiling point $(\approx 70^{\circ} \, \text{F})$, vaporization of the oxidizer was significant, resulting more closely to a gas/liquid impingement. For nitrogen tetroxide temperature below the boiling point, liquid/liquid impingement was observed. Figure 5* shows a photograph of a gas/liquid impingement. The flash light intensity (30 million beam candles) completely masked out the flame. The short flash duration (1.0 μ second) stopped the motion of propellant droplets and ligaments. The presence of the oxidizer vapor was evidenced by the slight spreading of the jet, with the flow velocity mainly along the injector axis. The nitrogen tetroxide vapor attained a much higher velocity as compared with that of liquid hydrazine. The resulting vapor stream carried the hydrazine droplets along with it, confining the hydrazine droplets mainly on the fuel side. This corresponded to a "separated" flow. Figure 6 is the photograph of the identical flow situation. The photograph was taken with 0.5 second exposure time without the flash and thus showed the combustion flames of the impingement region. The flame light indicated that the combustion occurred mainly on the hydrazine side, similarly indicating a "separated" flow.

^{*}Color photographs are presented in film supplement C-273 to this report, which is available on load; a request card and a description of the film are included at the back of the report.

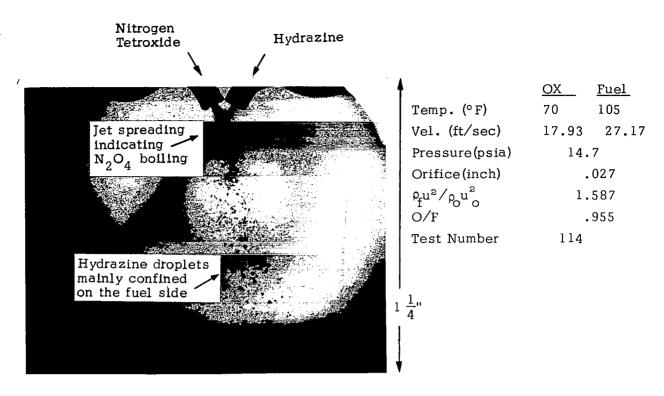


Figure 5. Stream Separation With Two Phase $\mathrm{N_2O_4}$ Stream.

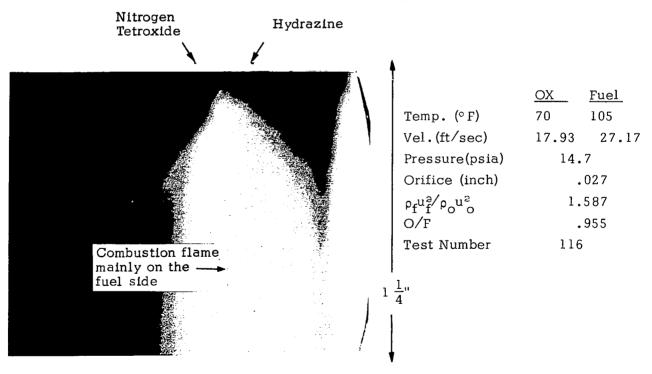


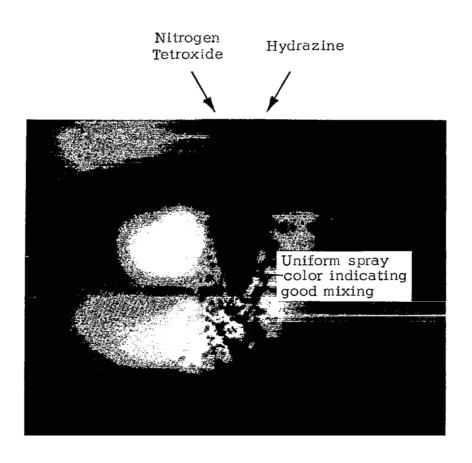
Figure 6. Time Exposure Photograph of Stream Separation with Two Phase $\mathrm{N_2O_4}$ Stream.

For nitrogen tetroxide temperature below 70°F, liquid/liquid stream impingement was observed. The color of the resulting spray was very uniform showing that droplets and ligaments of liquid fuel and liquid oxidizer were intermixed. Figure 7 is an illustration of the stream mixing case.

Figure 8 shows the result of the impingement study under atmospheric pressure conditions. The data are presented in terms of D/V (the ratio of orifice diameter to average propellant velocity) and the temperature of nitrogen tetroxide at the injectors. The data indicated that for nitrogen tetroxide temperature above approximately 70°F, very poor liquid mixing occurred due to the vaporization of nitrogen tetroxide stream as discussed previously. For nitrogen tetroxide temperature below approximately 70°F, the resulting spray showed good mixing of the liquid propellants. Injector popping was also observed within these temperature ranges. The simultaneous occurrence of both popping and stream mixing will be discussed in more detail in the injector popping section of this report.

Effect of unequal propellant temperatures on stream mix/separation phenomena was also studied under atmospheric pressure conditions. The result presented in Figure 9 indicated that hydrazine temperature played no significant role. Influence of unequal propellant velocities was also found to be insignificant when approximately equal temperature streams of hydrazine and nitrogen tetroxide impinged at atmospheric pressures. Effects of additives were also evaluated and found to be unimportant. 1% and 5% of unsymmetric dimethylhydrazine, monomethylhydrazine and ammonium nitrate were respectively added to liquid hydrazine and no significant influence on stream mixing and separation was observed.

In conclusion, under atmospheric pressures, the temperature of nitrogen tetroxide streams played the dominant role with respect to stream mixing and separating phenomena. Stream separation was never observed when nitrogen tetroxide was below its boiling point. For nitrogen tetroxide above the boiling point, stream separation occurred due to vapor/liquid impingement. Effects of unequal stream velocities, unequal propellant temperatures, and various additives in hydrazine were found to be insignificant.



	<u>OX</u>	<u>Fuel</u>
Temperature (^O F)	54	54
Velocity (ft/sec)	11.85	14.49
Pressure (Psia)	14.7	
$\rho_f^{u^2}f^{\rho_0u^2}$	1.1	7
Orifice (inch)	.0	60
Run No.	136	

Figure 7. Example of Stream Mixing with Liquid/Liquid Impingement.



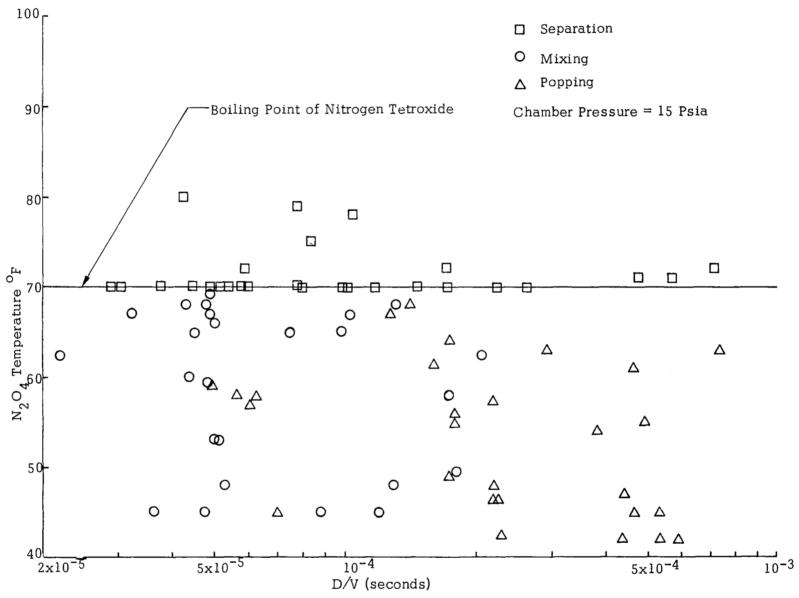


Figure 8. Nitrogen Tetroxide-Hydrazine Stream Impingement Data.

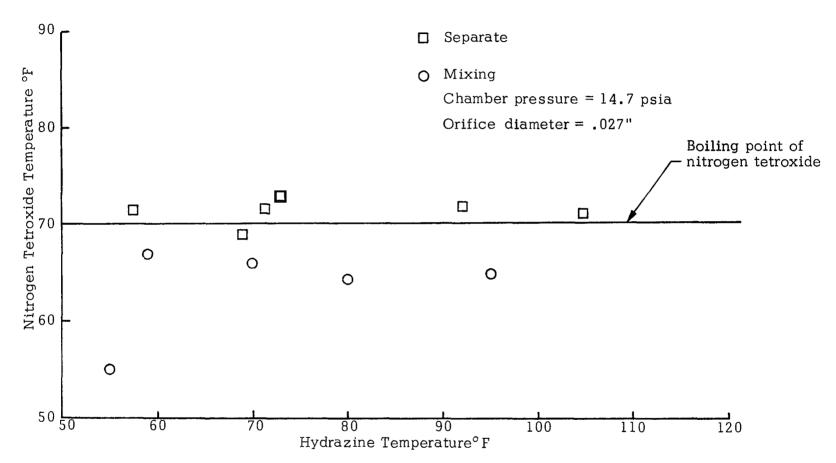


Figure 9. Effect of Propellant Temperature Difference on Stream Mixing and Separation. (.95 $< \frac{o_f u_f^2}{\rho_o u_o^2} < 1.05$)

Comparison With Previous Experimental Studies

Impingement experiments have been previously conducted (Ref. 6) using a different injector system. It consisted of an aluminum block with orifice slots on which a lucite observation window was bolted with a heavy retaining plate. The block was cut away at the impingement point to allow fan formation away from the lucite window. This injector design was different from the existing setup in two ways. First, the propellants were led to flow over the lucite plate so that the lucite burning could affect the photographic observations especially when photographs were obtained from the lucite view of the injector. Second, the impingement occurred at the immediate exit of the orifice, with no provision for jet free travel. The photographic observations of the impingement region used a 10μ second duration strobe back-lighting. Photographs of the combustion flames with 0.5 second exposure time without flash backlighting were also carried out.

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The experimental data previously obtained (Fig. 10) in general, agreed very well with that shown in Figure 8. The limiting boundary line (Fig. 10) separating regions of separation with that of mixing and popping was indeed close to the boiling point of nitrogen tetroxide. Based on the mixing and separation criterion previously defined (Ref. 6), stream separation was attributed to tremendous gas release associated with chemical reactions. The present study suggested that stream separation was mainly due to the boiling of the nitrogen tetroxide stream. To resolve this disagreement, experiments were performed using the slot injector with the lucite plate. High speed motion pictures, with a 2.0 μ second exposure time per frame were used to observe the impingement region. A significant amount of oxidizer gas was observed under previously separated conditions, indicating that the separation phenomena could be caused by the boiling of nitrogen tetroxide rather than due to reactions at the interface between propellants.

In summary, the previously proposed mixing/separation criteria of Reference 6 in the temperature ranges where propellant boiling was significant, has to be modified. Furthermore, under the temperature and pressure conditions where separation was observed, further experimental work indicated that separation was attributed to the boiling of nitrogen tetroxide resulting in gas/liquid impingement and thus produced very poor propellant mixing.

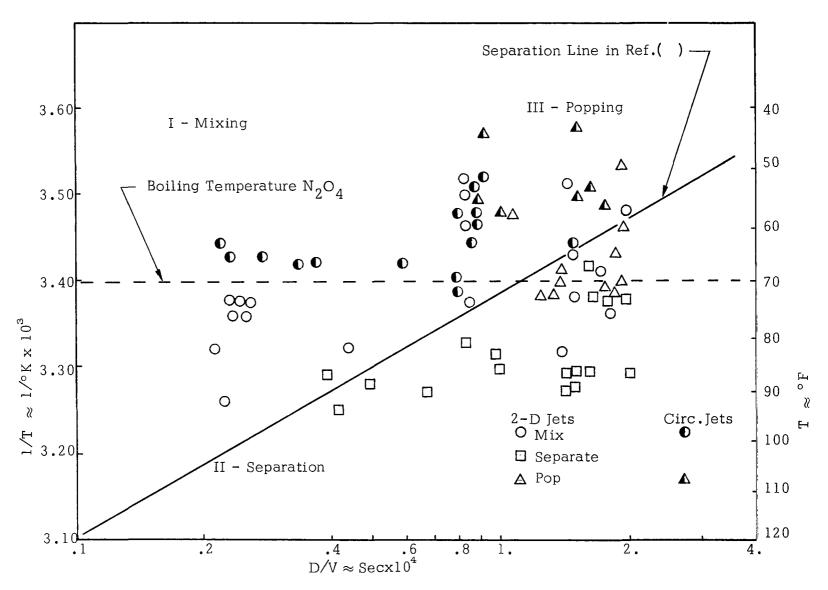


Figure 10. Impinging Jet Separation Data for N_2O_4/N_2H_4 Streams, From Ref. (6).

STREAM MIXING/SEPARATION RESULTS AT ELEVATED CHAMBER PRESSURES

Stream mixing and separation phenomena at elevated pressures were studied by impinging liquid hydrazine and nitrogen tetroxide within a pressure test chamber (Figure 1). The experimental conditions were as follows:

Chamber pressure : 15 Psia - 500 Psia

Propellant Temperature

Hydrazine : $40^{\circ}F - 140^{\circ}F$

Nitrogen Tetroxide : $40^{\circ}F - 140^{\circ}F$

Propellant Velocity

Hydrazine : 15 ft/sec - 90 ft/sec

Nitrogen Tetroxide : 15 ft/sec - 80 ft/sec

Orifice Diameters : .040" - .055"

Nozzle Diameters : 1.5", .600", .400", .283",

179", and .128".

Single exposure color photography and high speed color motion pictures were employed during this phase of study. For single exposure photography a high intensity, microsecond duration flash was used. A high intensity tungston lamp was used to provide proper lighting for high speed motion picture observations. With a 1/100 exposure ratio shutter, a two microsecond exposure time was obtained with a framing speed of 5000 per second. The high intensity lighting technique completely masked out the flame light, and the microsecond exposure time, $(1.0\mu \text{ second for single})$ exposure and 2.0μ second for high speed motion pictures) stopped the liquid stream and droplet motion, thus allowing clear observation of liquid hydrazine and liquid nitrogen tetroxide. The colors of the two impinging streams were distinguishable; hydrazine being colorless and nitrogen tetroxide being darkish brown. Mixing and separation phenomena were determined based on the color distinction of the two liquid propellants within the spray. For sprays with uniform brownish color, the impingement phenomenon was defined as mixed. Whenever color variations within the sprays were observed, with colorless liquid hydrazine along the fuel side of the spray, the impingement phenomenon was defined as "separated."

Figure 11 is a photograph of a mixed spray resulting from impinging liquid hydrazine stream with that of nitrogen tetroxide. The color of the spray was very uniform, indicating good intermixing of droplets and ligaments of two liquid propellants. Figure 12 is an illustration of poor propellant mixing. The color variation within the spray indicated that most of the colorless liquid hydrazine remained on the fuel side while brownish liquid nitrogen tetroxide mainly remained on the oxide side.

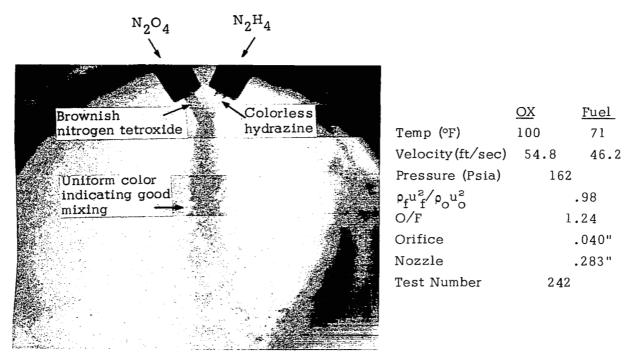


Figure 11. Typical Spray Photograph for Stream Mixing at High Pressure.

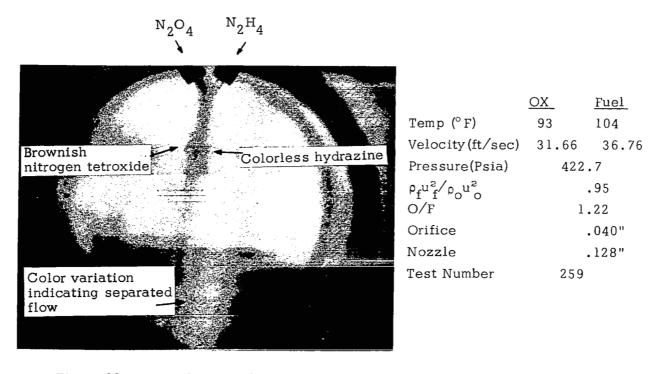


Figure 12. Typical Spray Photograph for Stream Separation at High Pressures.

This is an example of poor intermixing of droplets and ligaments of two liquid propellants and is defined as "separated" flow. High speed motion pictures of the sprays with very similar flow conditions as those shown by Figure 11 and Figure 12 were also recorded. It was difficult to distinguish the propellant colors since a high intensity tungston filament lamp was used to provide the backlighting. The motion pictures did show one distinct feature. Brownish oxidizer vapor was found on both sides of the resulting spray in the stream mixed case while oxidizer vapor was found to be more noticeable on the nitrogen tetroxide side of the spray in the stream separated case.

At low chamber pressures (under 60 Psia) vaporization of nitrogen tetroxide was again found to play a significant role. Separated flows due to gas/liquid impingement were observed. As evidenced by Figure 5 and Figure 12, the sprays of the two "separated" flows were significantly different.

Stream mixing/separation results at elevated chamber pressures are shown in Figure 13. Test data presented were only those with chamber pressure fluctuations less than 5%. In terms of average initial propellant temperatures and chamber pressures, two different separated flow regions were defined (shaded regions in Fig. 13). Separated flows were observed for chamber pressures above approximately 230 Psia. The color photographs of these sprays definitely showed a color variation with liquid hydrazine appearing on the fuel side of the spray. The effects of initial propellant temperature and injection velocities were found to be not significant, (similar to Fig. 12). At lower chamber pressure (less than ≈ 230 Psia) and for nitrogen tetroxide below its boiling point, sprays with very uniform brownish color were observed. These were the mixed flows. At chamber pressures below 60 Psia and for initial nitrogen tetroxide temperature above the boiling point, separated flows as a result of gas/ liquid impingement, were observed. The data agreed very well with the nitrogen tetroxide vapor pressure curves. For chamber pressure at 55 Psia, gas/liquid impingement was observed for nitrogen tetroxide temperature above 133°F, resulting in a spray pattern similar to that shown in Figure 5. The effect of different propellant velocities was found to be not significant.

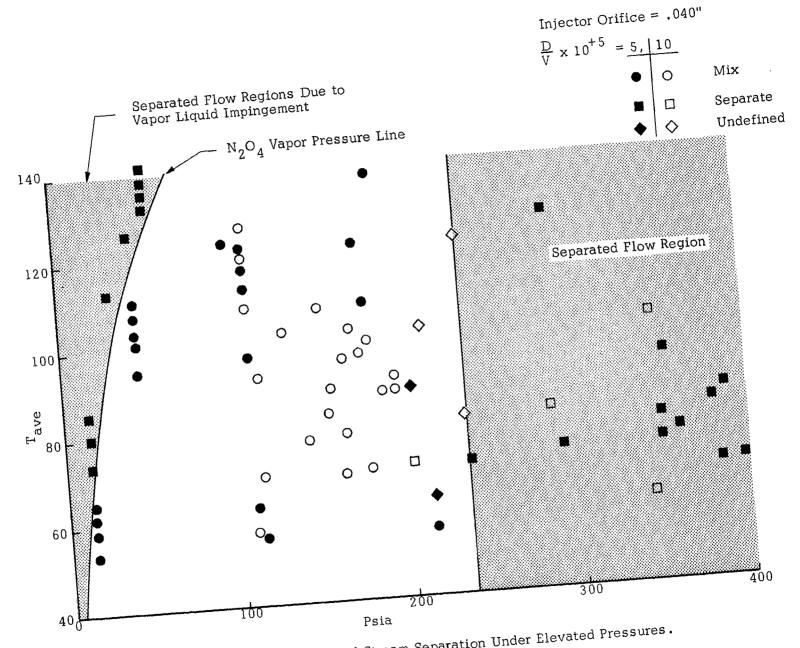


Figure 13. Region of Stream Mixing and Stream Separation Under Elevated Pressures.

COMBUSTOR PRESSURE POPPING

Transient high amplitude pressure disturbances, commonly referred to as "pops" have been observed during steady state rocket engine operation. The term "pop" has also been used to describe altitude ignition spikes and shutdown spikes. In the present discussion, the term popping will be used to describe high pressure disturbances associated with a single doublet injector element during steady state engine operation conditions.

Popping was studied under atmospheric and elevated pressures with impinging liquid streams of hydrazine and nitrogen tetroxide. The occurrence of popping was particularly easy to distinguish when impingement experiments were conducted in the open atmosphere. This was evidenced by the loud noise associated with these explosions. The noise of these explosions was recorded on the oscillograph via a microphone. Popping studies under elevated pressures were investigated by impinging propellant streams within the combustion chamber as shown in Figure 1. High speed motion picture technique was used to observe the popping. Chamber pressures were simultaneously recorded on the oscillograph and on the movie film via an oscilloscope. The occurrence of popping was determined based on motion picture observations and the related pressure spikes.

Experimental results showed that the two dominant parameters governing the occurrence of popping are the orifice injector diameter and the chamber pressure. Under atmospheric pressure with nitrogen tetroxide temperature below its boiling point, test results showed that popping was never observed when .027" diameter injector was used. Popping was observed infrequently when .040" diameter was used. Popping was always observed when .055" and .060" diameter orifices were used.

A typical occurrence of popping is illustrated in Figure 14. The photographs were reproduced from motion picture recordings. The film speed was 5280 frames per second. With the addition of a 1/100 exposure ratio shutter, the exposure time was approximately 2.0μ seconds. The experimental conditions were:

Chamber Pressure : 100 Psia

Propellant Temperature

Hydrazine : 71.5°F Nitrogen Tetroxide : 68.0°F

Propellant Velocity

Hydrazine : 52.8 ft/sec

Nitrogen Tetroxide : 46.5 ft/sec

Orifice Diameter : .060"



t = 0.0 sec., $P_C = 100 \text{ psia}$



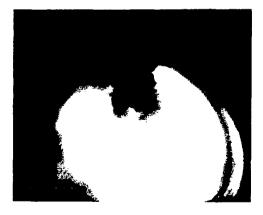
 $t = 1326\mu \text{ sec., } P_C = 190 \text{ psia}$



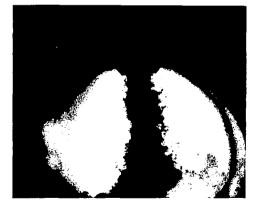
 $t = 189\mu \text{ sec., } P_{_{\scriptsize C}} > 475 \text{ psia}$



 $t = 1894 \mu \text{ sec., } P_{_{\mathbf{C}}} = 175 \text{ psia}$



 $t = 757\mu \text{ sec., } P_{_{\mathbf{C}}} > 475 \text{ psia}$



 $t = 3409\mu \text{ sec., } P_C = 100 \text{ psia}$

Figure 14. Hydrazine/Nitrogen Tetroxide Injector Popping Sequence.
Test Number 479. Steady State Chamber Pressure =
100 psia.

The pressure spike associated with the popping at a time of 189μ seconds in Figure 14, was in excess of 475 Psia. The entire popping sequence took approximately 3000μ seconds.

By increasing the chamber pressure to approximately 195 Psia (by replacing the chamber nozzle from .600" to .400") high pressure spikes were no longer recorded on the oscillograph results. The corresponding high speed motion picture recordings also did not show the violent explosion disturbance of popping. Only intermittant stream blow aparts were observed. These blow aparts did not produce any appreciable pressure spikes, but did produce chamber pressure fluctuations of approximately 10%. Figure 15 is a typical sequence of this blow apart phenomenon. The experimental conditions were:

Chamber Pressure : 195 Psia

Propellant Temperature

Hydrazine : 70° F

Nitrogen Tetroxide: 68°F

Propellant Velocity

Hydrazine : 51.0 ft/sec

Nitrogen Tetroxide : 46.5 ft/sec

Orifice Diameter : .060"

These tests showed that poppings which were observed with chamber pressure at 100 Psia were no longer observed with chamber pressure at 195 Psia. Color movies of popping (Fig. 14) and absence of popping (Fig. 15) are shown in the film supplement to this report.

The effect of chamber pressure can be briefly summarized as follows: at higher chamber pressures (Pc > 40 Psia) popping was not observed for both .027" and .040" diameter orifices. For .055" and .060" diameter orifices, popping was rarely observed when chamber pressures were above 185 Psia. The effect of chamber pressure and orifice diameter on the occurrence of popping is shown in Figure 16. It has been observed that popping was more likely to occur for large orifice diameter at lower chamber pressures and less likely to occur for smaller orifices and at higher chamber pressures. Both propellant temperature and velocity were found to be not important.

Popping frequency and average pressure spikes were obtained from the oscillograph data. These results are shown in Figure 17 and 18. Although the data were not reproducible, it did show the effect of pressure. Both the popping frequency and the magnitude of the pressure spikes are higher at lower chamber pressures.

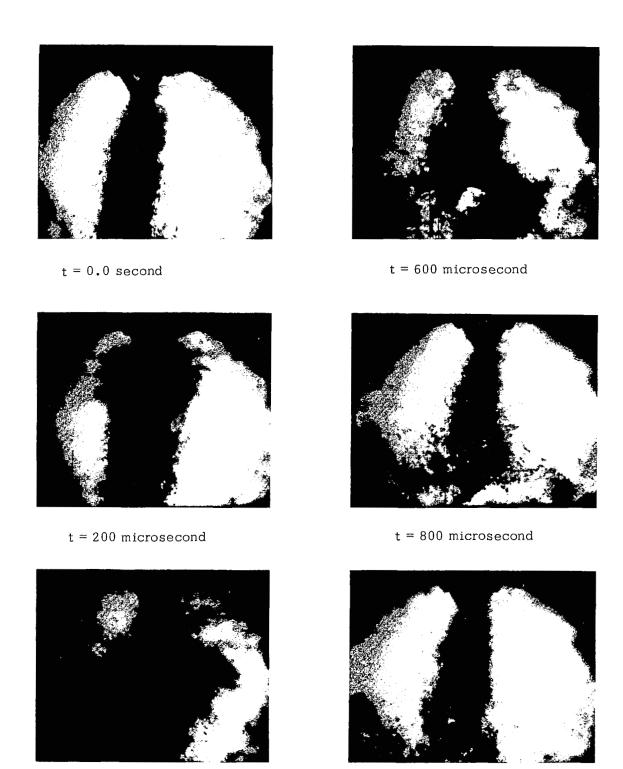


Figure 15. Hydrazine/Nitrogen Tetroxide Blow Apart Phenomena. (Test number 386, Steady State Chamber Pressure = 195 psia).

 $t - 1000 \; \text{microsecond}$

t - 400 microsecond

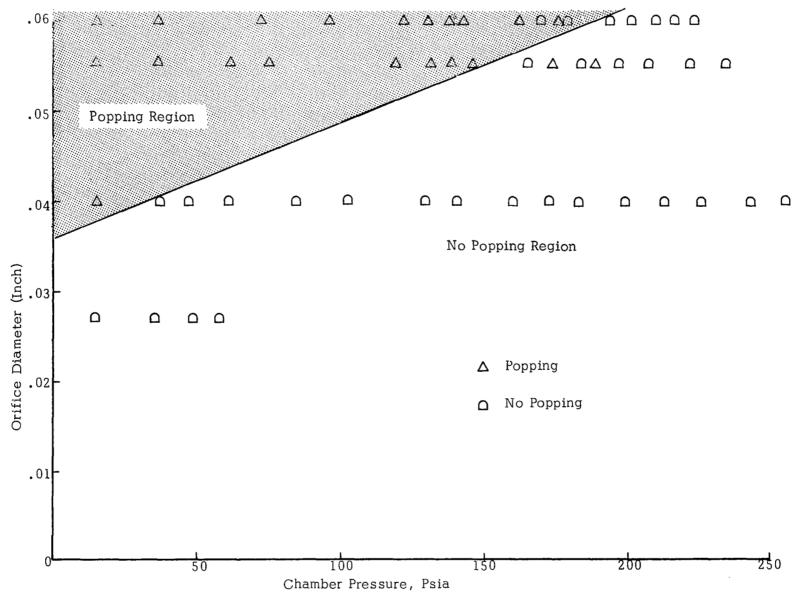
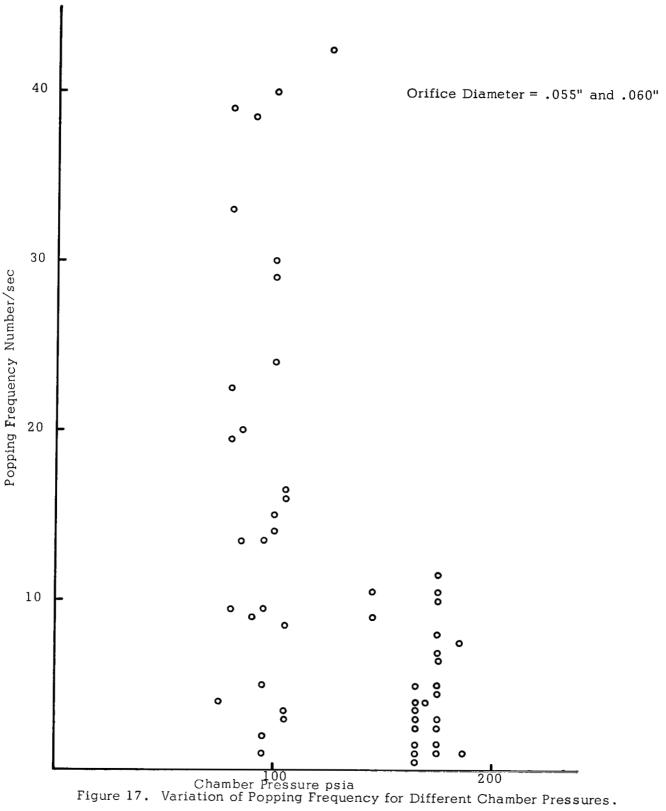


Figure 16. Hydrazine/Nitrogen Tetroxide Popping Regions.



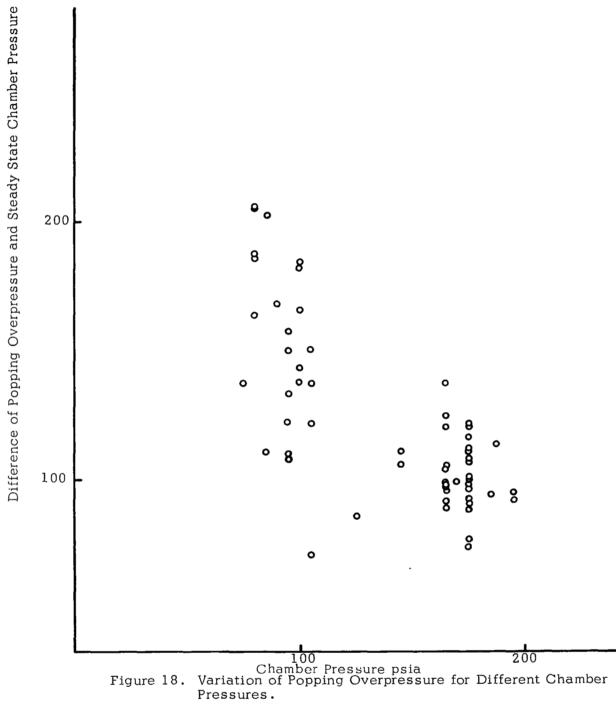


Figure 18.

RESULTS AND CONCLUSIONS

Both single exposure photography and high speed motion pictures were used to study three different combustion phenomena associated with impinging hypergolic propellant streams: combustor pressure poppings, stream mixing, and stream separation.

High speed motion pictures showed that popping is a violent explosion of the liquid propellant sprays. Chamber pressures were also recorded and were correlated with the photographically observed explosions. Pressure spikes of several hundred Psia were often observed resulting from these explosions. The occurrence of popping was found to depend significantly on chamber pressure and injector diameters. Using .055" and .060" injectors, experimental data showed that popping was almost absent for chamber pressures above 185 Psia, and was found to occur very frequently at lower chamber pressures. Popping was never observed using the smaller .027" orifice, and was observed only infrequently at low chamber pressures when .040" orifice was used. Popping frequency and average pressure spikes were obtained from the oscillograph data. Both the popping frequency and the magnitude of the pressure spikes are higher at lower chamber pressures. To summarize, the test results showed that popping was more likely to occur for larger orifice diameters at lower chamber pressures and steady state chamber operations were observed for smaller orifices at higher operating chamber pressures.

Stream mixing and separated flows were studied under steady state chamber operation conditions. Two distinctly different separated flows were observed. At low chamber pressures and for nitrogen tetroxide temperatures above the boiling point, vaporization of the oxidizer was significant, resulting in a gas/liquid impingement. The resulting spray pattern showed that hydrazine droplets were mainly confined on the fuel side. Photographs of the flame also showed that combustion mainly occurred on the fuel side indicating a separated flow. Separated flows were also observed with liquid/liquid impingement at higher chamber pressures (above 230 Psia). The resulting liquid sprays definitely showed color variations with colorless hydrazine along the fuel side of the spray and brownish liquid nitrogen tetroxide on the oxide side. This indicated very poor intermixing of droplets and ligaments of two liquid propellants. Stream mixing flows were observed at lower chamber pressures (less than 230 Psia) and with nitrogen tetroxide temperatures below the boiling point. The sprays resulting from liquid/liquid impingement showed a very uniform brownish color, indicating good intermixing of droplets and ligaments of the two propellants. To summarize, the test results showed that chamber pressure played a very significant role. Separated flows were observed at high chamber pressures and mixed flows were observed at lower chamber pressures. In addition, nitrogen tetroxide temperature was also found to be important. For nitrogen tetroxide temperature above the boiling point, gas/liquid impingement was observed resulting in separated flows.

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APPENDIX A

NOMENCLATURE FOR TABULAR TEST DATA

F	=	Fuel
0	=	Oxidizer
W	=	Propellant Flow Rate (lb/sec)
T	=	Propellant Injection Temperature (°F)
V	=	Propellant Injection Velocity (ft/sec)
M	==	Momentum of Propellant Stream (WV=16 ft/sec2)
\bar{v}_a	=	Average Injection Velocity = $(W_f V_f + W_o V_o)/W_o + W$)
$\frac{M_f}{M_o}$	=	Momentum Ratio of Fuel and Oxidizer Streams
D	=	Injector Orifice Diameter (equal fuel and oxidizer orifices)
$\mathbf{\bar{T}_{a}}$	=	Average Injection Temperature= $(W_f T_f + W_o T_o)/W_f + W_o)$
$\mathrm{D}/\mathrm{\bar{V}}_\mathrm{a}$	=	Ratio of Orifice Diameter to Average Injection Velocity
P	=	Chamber Pressure.

Test W T V M V O/T M/MO D T D/V P Comments Test W T V M V NO. 15 F .0087 70 15.05 1305 29.17 .942 1.621 .027 70 7.71x10 14.1 Ox boiling; critice out of alignment Ox box boiling; critice out of alignment Ox boiling; critice out of al	0/F	M _f /M _c	Ŋυ	. T.	D/V.	i P	
15 U 0.0082 70 21.94 .188 29.17 .942 1.621 .027 70 7.71x10 14.1 OX bottom; orifice out of alignment 102 O.0065 78 18.21 .1184	1.10		Ψ	7	 	- `	Comments
	1	1.20	ـــــــــــــــــــــــــــــــــــــــ	80	<u> </u>	_!	Time exposure photo
28 F .0156 70 74.97 1.3945 CO FD .020 20 20 20 20 20 20 20 20 20 20 20 20	1.89	.407	.027	82.5	5.75x10	14.	Separated flow due to Ox boiling
For F 011 70 45 9 505 95 30.25 .2269 38.11	2.01	.405	-	82.5	5.91×10	5 "	Separated flow due to Ox boiling
O 0195 68 54.9 1.076 51.7 1.78 469 027 68.74.3×10 ⁻³ mixed. O 0151 70 42.02 .6345	-		+	╁╌	-5	╁	Separated flow due to
61 F .010 70 40.34 .4034 41.3 1.50 .641 .027 70 5.4x10 ⁻⁵ 14.7 Separated flow due to Ox boiling.	1.18	1.06	ļ <u>.</u>	-	5 8.0x10 ⁻⁵		Ox boiling
F .0076 72234 37.6 1.95 .384 .027 72 .5.9x10 ⁻⁵ 14.7 Separated flow due to Ox boiling.	1.38	.76	"	65	9,8x10 ⁻⁵		Mixed flow
F .0082 65 33.2 .272 30	1.16	1.07	Ţ	65	7.5×10 ⁻⁵	-	Mixed flow
O .0146 67 40.90 .597	+	1	+	+-		5 .	Separated flow due to
64 F.0132 68 53.23 .729 49.5 1.21 1.016 .027 66.2 4.5x10 ⁻⁵ Mixed flow 113 O.0086 70 23.9 .2043	1.04			-	7.76×10	1	Ox boiling
F 0195 70 78.45 1.53 74.0 1.29 .86 .027 70 3.04x10 ⁻⁵ 14.7 Separated flow due to 114 F .0067 105 27.17 .1820 22.65	.955	1.587	` "	87.4	4 9.9x10 ⁻⁵	-	Separated flow due to Ox boiling
P no62 70 75.21 78	.95	551.58	1.	87.	4 9.97x15	5 .	Time exposure photo
O 0069 70 19.55 .13 Ox boiling. U 1048 70 17.93 .1147	+-	1 . 50	+	-	1.04×10	1.	Time exposure photo
80 F 10134 68 54.35 .728 47. 1.09 1.220 .027 68 4.78x10 ⁻³ 14.7 Mixed flow.		6 1.58	ļ	┿	-	_	Time exposure pioto
81 F 0132 70 53.23 .704 43.8 .909 1.747 .027 70 5.13x10 ⁻⁵ 14.7 separated flow due to Dy bolling Or	1.07	7 1.28	.060	75	3.74×10 ⁻¹	5	Time exposure photo
O 0120 70 33.52 .403 F 0133 70 52.0 .582 46.1 .02 1.696 .027 70 4.88x10 ⁻⁵ 14.7 Separated flow due to	1.23	3 .95		75	14.6x10 ⁻⁵	5 ,.	Separated flow due to Ox boiling
82 O 0136 70 39.5 .402	1.12	2 1.15	-	75	11.73x10	-5 -	Time exposure photo
83 70 52.0 .092 46.1 1.073 1.195 027 70 4.88x10 14.1 Ox boiling. 120 0.0051 70 16.85 .0866	1.1	2 1:13	╀-	,,,	ļ		Time Exposure photo
10 10 18 70 47.6 .562 42.8 1.02 1.08 027 70 5.25x10 ⁻⁵ 14.7 Separated flow dui to boilting. 121 F 0118 80 47.9 .565 41.70 85 0 0136 70 38.1 .519 42.8 1.02 1.08 027 70 5.25x10 ⁻⁵ 14.7 Separated flow dui to boilting.	1.09	9 1.21	.060	75	5.39x10	14.	Time exposure photo
O 1050 68 124.4 . 1475 122 F 0227 80 92.5 2.1 79.74	1.06	6 1.28	-	75	2.82×10	-5	Time exposure photo
O 0057 67 18.6 .125 21.3 1.10 1.16 .0216	† 	1	†	1	ļ	·s -	Separated flow due to
90 0 0199 67 53.2 1.011 /2 .613 2.177 027 07 3.2410 4.17 mixed 100 0 0251 75 13.19 .349 13.99	1,16	+	╫-	75	1		Ox boiling Separated flow due to
F 0233 70 94.3 2.20 73.8 .815 2.177 " 70 3.05x10 ⁻⁵ 4.7 Ox boiling 124 O 0150 71 8.34 .125 10.45	.95	5 1.06	<u> </u>	71.	5 4.784x10		Ox boiling
F 0233 70 92 .208 78 1.02 1.38 " 70 2.89x10 ⁻⁵ " Separated flow due to Ox boiling 125 F 0135 72 10.65 .144 8.77	.89	9 1.81		71.4	4 5.70x10	4 -	Separated flow due to Ox boiling
O 0190 70 05 1.31 F 0108 72 8.55 092	.90	0 1.77	┧.	72	7,12×10	-4 -	Separated flow due to Ox boiling
93 O 0154 60 43.4 .668 51.25 1.04 1.31 60 4.39x10 Mixed flow O .0097 72 5.38 .052	+	┼	╀	╄-	 	.+-	
94 F 0139 45 56.0 .778 63.23 1.31 .83 " 45 4.75x10 ⁻⁵ " Mixed flow 127 P 0.0078 72 4.34 .0339 4.94	1.10	0 1.17	<u> </u>	72	1.0x10 ⁻³	<u>`</u>	Separated flow due to Ox boiling
95 F 0164 45 66.5 1.09 63.23 1.31 .83 " 45 3.62x10 ⁻⁵ " Mixed flow 128 F .0208 50 16.46 .342 14.54	1.11	1 1.15	-	54	3.44×10	٠ ١	Popping
0 02:6 45 60.8 1.313	.90	6 1.56	1-	56.	8 4.39x10	4 .	Popping
97 O 0078 45 22.0 .172 23.3 1,00 1.23 3 5.02.10 Pror photography O .0165 42 9.13 .150	+	┼	+-	+		4 -	
98 F 3072 45 29.3 .2103 25.72 1.11 1.17 " 45 8.75x10 ⁻⁵ " Mixed flow O 0154 42 8.56 .132 9.33	1.18	8 1.01	1_	56.	45.36x10	1	Popping
F 0158 80 64.3 1.016 52.84 .89 1.80 " 80 4.25x10-5 " Separated flow due to Ox boiling F 0113 52 8.95 101 8.40	1.2	7 .88	-	56.	8 5.95×10	4 -	Popping
F 0082 80 33.0 .2706 35.00 94.2 20 17.2 9.3710-5 Separated flow due to 132 F 0100 52 7.90 0.779 9.35	1.83	3 .425	1.	56.	8 5.35x10	4 -	Popping
0 0069 75 19.61 .1353 Separated flow due to	1	1					
101 F 0089 83 36.14 .3216 28.92 .80 2.28 " 80 7.78x10 ⁻⁵ " Ox boiling							

Test W T V M V O/F Mf/Mo D T	D/V P C	comments	Tes	it	w	т	v	м	v _a	O/F	M _f /M _o	D	Ta	D/Va	P	Comments
No. 11 85 177			No. 180	F	0129	80.5	23.74	.3062	21.07	1.55	1.07	-	1	1.57x10 ⁻⁴	165	Mixed flow
0.0178 45 9.88 .176 10.76 1.18 1.0	4.65×10-4 ··	Popping		100	0149	76	19.14	.2852				_	-			
134 F 0168 50 13.28 .223 11.30 1.0 1.423 .060 53.8 4	4.43×10 ⁻⁴ 14.7	Popping	181		.0137		17.61	.2412	18.86	1.234	.94		78.3	1.77x10 ⁻⁴	140	Mixed flow
	7.39x10 ⁻⁴ "	Popping	182		0106	87	19.40	.2049 .3086	19.67	1.47	.66	-	82	1.70×10 ⁻⁴	259	Difficult to define
2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	3.83×10 ⁻³ "	Popping	183	_	0158	93	29,10	.4598	24.27	.943	1.61	-	99 7	1.37x10 ⁻⁴	157	Mixed flow
0.0214 54 11.65 .235		Topping		-	0149	100.5	19,15	.2853				_	-1			
137 F 0133 33 12:13 103 10.17 1.04 1.54 " 54.8	4.92xi0 ⁻⁴ "	Popping	184	1-	0135	95	17.36	.0997	15.85	1.83	.425		96.7	2.10x10	119	Mixed flow
139 F 0153 61 12.07 .185 10.77 1.13 1.11 " 50.84	4.64×10-4 "	Popping	185		0096		17.62	.1691	17.42	1.41	.721	•	102.	1.91×104	131	Mixed flow
140 F 0253 63 19.98 .505 16.81 .98 1.47 " 62.82	2.97×10 ⁻⁴ "	Popping	186	F	0129	91	23.74	.3069	22,58	1.31	.837	040	82.6	1.48×10 ⁻⁴	155	Mixed flow
0 0249 63 13.83 .344 141 F 0139 70 25.53 .355 23.37 1.20 .986 .040 68	1.43×10 ⁻⁴ ; "	Popping	-	1	0169	_	21,70	.3667	26.76	1.41	.723	├	90	1.25×10 ⁻⁴	194	Mixed flow
0.0167 66 21.57 .360			187	0	0207		26.55	.5496	20.70	1.41	./23	L		1.23210	134	Maked How
142 F 0113 70 20.81 .235 19.20 1.22 .951 1 69.81	1.74×10 ⁻⁴	Separated flow due to Ox boiling	188		0160	90 85.5	29.36	.5080	27.46	1.24	.92	"	87.3	1.22×10 ⁻⁴	194	Mixed flow
143 F 0094 70 17.23 .161 14.96 1.07 1.25 . 69.8 2	2.23×10 ⁻⁴ "	Separated flow due to Ox boiling	191	F	.0136	102.5	25.02	.3408	24.88	1.42	.713	"	98.8	1.34×10 ⁻⁴	170	Mixed flow
F 0076 70 14.04 .107 12.71 1.18 1.03 " 59.8 2	2.62×10 ⁻⁴	Separated flow due to	-	F	.0147	113	27.06	.3978	25.35	1.27	.886	ļ	107.	51.32×10 ⁻⁴	152	Mixed flow
144 O 0090 70 11.62 .104	1 1	Ox boiling	192	10	0167	103.5	30.76	.4488			_	├	-			
010160 72 20.68 331	1.71×10 ⁻⁴ "	Separated flow due to ! Ox boiling	193	0	.0191	93.5	24.51	.4681	27.42	1.14	1.097	Ľ.	95.9	1.22×10 ⁻⁴	174	Mixed flow
149 F 0151 68 27.8 .420 25.51 1.21 .97 .040 68	1.31×10 ⁻⁴ 14.7	Mixed	194	_	0164	104	30.12	.4939	26.74	1.13	1.12	"	101.	1.25×10-4	169	Mixed flow
150 F 0154 59 28.34 .437 25.97 1.20 .98 " 63.8	8 1.28×10 ⁻⁴ "	Popping	196	F	.0172	94	31,66	.5445	27.67	1.09	1.21		88	1.204×10-4	194	Mixed flow
O 0186 67 24.00 446 5 19.15 268 20.56 1.22 1.29 " 63.1	11.62×10 ⁻⁴	Popping	197	1-1	.0172		31.66	.5445	25.05	.72	2.79	-	87.4	1.331×104	187	Mixed flow
Q 40127 61.5 16.34 .207	:			1 -	0123	_	15.83	.1947				-	 	4	101	Mixed flow
152 P 0104 63 19 15 199 16.31 1.01 1.40 62.8 O 0105 62.5 13.53 142	8 2.04×10 ⁻⁴ "	Mixed	198	0	0184	90.5	23.61	.4344	25.14	1:25	.92		94.7	1.33×10	164	Mixed flow
164 C.0127 68.5 16.39 .2081 19.07 1.068 1.25 " 71.6	6 1.75×10 ⁻⁴ 149	Mixed flow	199		.0176	92.5	32.42	.570	27.73	1.03	1.36		100	1.20×10-4	181	Mixed flow
165 F 0126 73 23.23 : .2927	2 1.43×10 ⁻⁴ 189	Mixed flow	200	F	0172		31.66	.5445	27.23	1.04	1.324	.040	70	1.224×10	203	Appeared to be mixed flow
F .0126 73 23.23 2927			201	l r	0179	77	31.66	.5445	26.25	.92	11.68	1.	69.6	1.27×10 ⁻⁴	178	Mixed flow
166 O 0186 66 24.12 .4486 23.76 1.47 .652 69.7				10	0159	$\overline{}$	20.42	.3247		1	1	-	 		+	·
167 F 0135 59 24.76 .3343 23.88 1.35, .79 " 51.9	9 ₁ 1.39×10 ⁻⁴ 189	Mixed flow	204	. —	.0194	63.	5 35.74 5 30.64	.693	32.92	1.23	.95	"	56.	7.1.01x10 ⁻⁴	110	Mixed flow
	1.324×10 ⁻⁴ 209	Difficult to define	205	, —	0189	84	34.72	.656	33.83	1.36	1.36	· "	79.2	9.85×10 ⁻⁵	115	Mixed flow
170 F 0082 57.5 15.06 .1235 15.13 1.45 .683 " 46.7	7,2,203×10 ⁴ ,111	Mixed flow	206	r	.0257	_	33.19 5 29.87	.853		-		+		·4	1	Mixed flow
O.0119 40. 15.19 1.1807	<u> </u>		 	0	.0259		5 33.44 5 33.70	.8661	32.06	1.59	1.59	+-	-		;	ļ
O 0180 42.5 23.0 1.4140 24.58 1.25 .923 49.2	191 1.356×10 ⁻⁴	Mixed flow	207	0	.0226	109.	5 29.10	.6577	31.16	1.23	1.23		108	1.069×10	110	Mixed flow
174 F 0145 64 26.55 .3850 24.25 1.20 .99 .040 58.2	2 1.374×10 ⁻⁴ 172	Mixed flow	208		.0217	108	39.83	.8643	33.86	.99	1.44		104.	6, 9.8×10 ⁻⁵	110	Mixed flow
175 F .0145 65 26.55 .3850 24.25 1.20 .99 " 60.4	4 1.374×10 4 187	Mixed flow	209	9 F	.0189	110	34.72	.6562	31.66	1.19	.99	<u> </u>	106.	61.05×10	110	Mixed flow
O 0174 57 22.34 .3887 179 F 0126 62.5 23.23 .2926 22.07 1.30 .842 " 58.6	6 1.51×10 ⁻⁴ 163	Mixed flow	<u> </u>	F	.0226	104.	37.02	.6577	22.00	1,		-	-	<u> </u>		Mixed flow
O 0164 56 21.19 .3475 22.07 1.30 .842 58.6	0.1.31810	Inneed How	210		.0226	121	29 10	.6577	32.03	1.12	1.13		p 15.	6 1.02×10	- 110	Mixed flow

Test	w	Т	T v	м	V _a	O/F	M _f /M	D	Ī.	D/V	ь	Comments	Te:	5!	w	Т	V	M	-	6.6	M _f /M _c	1 -	Τ-		-	Γ
No.	.0189		34.72		32.17	1.24	.93	 			-	Mixed flow	251		-	+ -		2.304	v _a _	O/F		D D			+	Comments
211 0	.0234	121	30.12	.7049	32.17	,		ļ	10.0		1		ll	0	.0436	61.	5 56.17	2.449	60.16	1.23	.94	<u> </u>	66.7	5.54×10 ⁻⁵	384	Separated
212 F	0189	117.5	34.72		32.18	1.24	.93	-	19,2	1.04×10-4	110	Mixed flow	252	Г	.0272	75.	5 50.04		47.16	1.28	.871	-	69.4	7.07×10 ⁻⁵	307	Not defined
215 F	.0368	118	-	2.490	61.17	1.17	.852	,040	112.4	5.45×10 ⁻⁵	110	Mixed flow	253		.0208		38.30		36.28	1.29	.853	-	71 4	0.10-10-5	F 02	Separated
217 F	.0432	101		2.404		-		<u> </u>						0	.0269	67.	5 34.72		30.25	1	.033		/1.4	9,19810	307	Separated
217 0	.0432	93	55.66	2.490	61.17	1.17	.852	"	96.5	5.45×10 ⁻⁵	110	Mixed flow	254	-	.0208		5 28.30 35.74		32.55	1.33	.594	١.	80.4	1.02×10 ⁻⁴	520	Separated
2 (9)	.0345	_		2.184	59.45	1.27	.7B7		122	5.61x10 ⁻⁵	110	Mixed flow	255	ſ	.0285	103	52.34	1.492	50.16	1.32	.818		105	6.65×10 ⁻⁵	84	Mixed
220 F	.0438	123.4	_	2.471		-				r ra5		Mixed flow	-	1_	.0376		5 48.51 5 65.10		-		<u> </u>	├-			⊢	
	.0416	116.5	53,61	2.230	60.20	1.13	.88	<u></u>			<u> </u>		256	0	.0447	126	56.17		60.11	1.26	,917		123	5.55×10 ⁻⁵	99.	Mixed
221 [.0368	110	58.72	2.490	62.72	1.24	808.	"	105.9	5.32×10 ⁻³	110	Mixed flow	257	F	.0361	120		2,396	60.79	1.20	.981	.040	121.9	5.48x10-5	99.	Mixed
222 F	.0368			_	60.80	1.16	.86	-	116.4	5.48×10 ⁻⁵	110	Mixed flow	350	-			5 56.17 36.76				<u> </u>	 -				
10	.0426	118.5		2.338		1111		_	<u> </u>				258	. 0	0245	93	31.66	.7757	33.95	, 1.22	.95	:	37.7	9.82×10 ⁻³	422.	7 Separated
	.0368			2.490	63.11	1.52	.798	"	1 16.5	5.28x10 ⁻⁵	110	Photograph of plume	259		0203		37.27		34.45	1.22	.843		96.3	9.68×10 ⁻⁵	461	Separated
231 F	.0361	65.5		2.396	58.36	1.1	1.18	040	72.6	5.71x10-5	350	Separated flow	260		0200				33.80	1.24	.917		RB 3	9.86×10-5	461	Separated
232 F	.0361	71.5	51.06		60.15	1 1 10	1.00		28 6	5 54v10 ⁻⁵	380	Separated flow			.0247			.788								
. 0	.0426	85	54.89		00.13	1							261	0	.0200	79	36.76		33.43	1.18	1.01	. •	32.9	9.97×10 ⁻⁵	115	Mixed
234 F	.0368	117.5	67.66		68.07	1.12	.883		117.3	4.897x10 ⁻⁵	353	Separated flow	262		0225		5 41.36		37.37	1.18	1.04		93.9	8.92×10 ⁻⁵	130	Mixed
235 F	.0368	102.5			58.61	1.05	.948		93.8	5.68x10 ⁻⁵	353	Separated flow	1	_ 0	.0263 .0285		33.96 5 52.34			<u>. </u>	-	<u> </u>	-			
	.0388		50.04	2.127									1	10	.0313	1 89.	5 40.34	1.263	46.05	1.1	1.18		92.6	7.24×10 ⁻⁵	149	Mixed
236 O	.0368	70.5	47.48		54.72	1.08	.924		79.4	6.09x10 ⁻³	346	Separated flow	264	F	0288	96.	5 52.98 5 45.95		49.10	1.24	.933	-	92.6	6.80×10 ⁻⁵	169	Mixed
237 F	.0368	93.2		2,490	51.44	1.18	.844	-	67.5	5.43×10 ⁻⁵	362	Separated flow	265	F	0334	92.		2.046	55.41	1.17	1.03		87.7	6.01x10 ⁻⁵	188	Mixed
	.0436	63 85	67.66	2.450	58.61	1 05	.95		74 0	5 7810-5	251	Separated flow	l——	1 -	0375	_	50.55	1.981								
1 10	.0368	65.5			30.61	1.03							266		0435	83		2.443	62.07	1.16	.862		B7.2	5.37x10 ⁻⁵	203	Mixed
23 9 F	.0146	82 72	25.80	.3912	24.44	1.19	1.0	"	76.4	1.36×10 ⁻⁴	145	Mixed	257	-	0368	88		2.489	62.44	1.22	.948	-	B3.7	5.34x10 ⁻⁵	388	Separated
240 F	.0368		67.66	2.490	66,17	1.37	.76		64.3	5.04x10 ⁻⁵	238	Undefined	┞─	F	0451	80.	57.44	1,792		_					-	
241 F	.0505	63 117	_	3.287 4.250		-	<u> </u>		_				268	-	0451	77.			57.88	1.44	.683		80.8	5.76x10 ⁻⁵	381	Separated
1 10	1.0574	114.5		2.585	71.55	1.56	.586		115.3	4.66×10	230	Mixed flow	269	$\overline{}$	0285	86 77	52.34		53.87	1.49	.669	"	80.4	6.19×10 ⁻⁵	353	Separated
242 F	.0375	126.5		2.585	61.46	1.14	1.11	-	136.8	5.42x10 ⁻⁵	176	Mixed flow	270	F	0389	129	71.48	2.781	60.27	.966	1.52	040	134.8	5.531×10	176	Mixed flow
-	.0375	122	68.93	2.585	63.42	1 125	1.11	-	119	5.42×10-5	176	Mixed flow		_	0376	137.6		2.133				_	_			
	.0425		-	2.337	61.47	1.133		_			176	Wived How	272		0437				59.1	1.28	.865		107.	5.64x10 ⁵	179	Mixed flow
244 0	.0352	53.5		1.199	52.85	.866	1.89	.040	52	6.3lx10 ⁻⁵	338	Separated flow	273		0341	123.5	_	3.133	59.94	1.31	.82		21.4	5.56×10 ⁻⁵	176	Mixed flow
246 5	.0334	40.6	61.27	2.046	58.65	1.31	.822		39.9	5.68×10 ⁻⁵	400	Separated flow	274		.0449 .0250	120.		1.149				_	-			
247 F	1.0439	39.4	56.68	2.488				-	_	- 41 . 4-S			-/4	0	0426	117	54.89	2.338	51.58	1.7	.49		17,9	6.46×10 ⁻⁵	176	Mixed flow
1 10	.0449	50.7	57.95	2.602	61.58	1.25	,915		55.3	5.41x10	215	Mixed flow	275		0278	117		1.419	46.21	1.17	1.03	-	14.9	7.21x10 ⁻⁵	138	Mixed flow
248 F	.0354	60	61.27	2.304	62.89	1.34	.792	"	55.4	5.3×10 ⁻⁵	115	Mixed flow	276	F	040	-70.	73.4		74.11	1.447	.679	Ι.	70	4.49x10 ⁻⁵	279	Flow undefined
-	0368	69.5		2,490	64.06	1.29	. 885		52 P	5.20x10 ⁻⁵	210	Mixed			0579		74.7	2.91				_	\vdash			
250 O	.0475	58	61.27	2.910	04.00	123			٥. ،	J.20X10	218	mixed	278	$\overline{}$	0471	70	_	2.86	66.47	1.187	.983	٠.	70	5.0x10 ^{-S}	237	Mixed flow
																				_ 1					-	

Test	- <u>-</u>		-		M _t /M _o	,—	-												14 (64)					
No. W	1	V M	V _a	O/F		D	T _o	D/V _a	P	Comments	Tes	. !	W	T	٧	M	v _a	O/F	M _f /M _o	D)	Ta	D/V _a	P	Comments
279 F .0397		73.4 7.914 66.12 3.39	69.27	1.292	1.16	"	70	4,81x10 ⁻⁵	250	Mixed flow	308	r o	.029	8 58.5 58	54.9 49.5	1.65	53.2	1.3	.86	4	58	6.25×10 ⁻⁵	6	Popping based on O-graph record
28: F .069		26.8 8.74 00.5 7.83	112.5	1.13	1.115		71.	2.96x10 ⁻⁵	100	Mixed flow	309		035	58.5	66.4 57.5	2.22	59.1	1.25	.86	.040	58	5.63x10 ⁻⁵	14.7	Popping based on O-graph record
292 F .069	72.5 12	6.5 8.73	117.2	1.54	.872	-	70	2.66x10 ⁻⁵	115	Mixed flow	310		037	77.5	67.5 57.5	3.04	68.7	1.25	1.17		73	4.85x10 ⁻⁵		Mixed flow
1281	97.5	10.05 37.5 5.16	97.7	1.22	.943	.040	<u> </u>	3.8×10 ⁻⁵	326	Separated flow	311	F	.037	66.	67.5	3.04	68.7	1.25	1.17		62	4.85x10 ⁻⁵	*	Mixed flow
101.065	84.2 74	93.1 4.65	83.5	1.16	1.06	 	-	3.38x10 ⁻⁵	282	Separated flow	312	r	.045	59.5 61	57.5	2.59							_	···
0.058	54 86	75.5 4.37		├		-	<u> </u>			<u> </u>	\vdash	O r	.049	66 55	63.8	3.13	65.5	1.25	.80	-	65	5.0×10 ⁻⁵		Mixed flow
7 074	54 86	62.8 3.08	69.7	1.16	1.06	ļ		4.78×10 ⁻⁵	228	Mixed flow	313	0	.049	53	63.8	3.13	65.5	1.25	.80		54	5.0x10 ⁻⁵		Mixed flow
400	54	62.8 2.13 51.3 2.00	55.5	1.14	1.06	"	69	5.9x10 ⁻⁵	192	Mixed flow	314	_	.037	55 53	67.7	3.00	64.7	1.29	.80	*	54	5.1x10 ⁻⁵		Mixed flow
287	86 55	51.06 1.37 44.1 1.49	46.8	1.25	.92	-	69	7.12x10 ⁻⁵	182	Mixed flow	316	F	.034	51 48	62.6	3.06	62.5	1.44	.70		48	5.3x10 ⁻⁵		Mixed flow
288 1 .018	57	51.8 0.93 45.9 1.56	47.9	1.89	.597	"	60	6.94×10 ⁻⁵	186	Mixed flow	317	F	029	51	54.9	1.59	47.4	1.45	.71	*	48	7.0x10 ⁻⁵		Popping based on O-graph record
289 F .028	55	53.2 1.48	50.5	1.32	.822	-,-	59	6.60×10 ⁻⁵	174	Mixed flow	318	o F	015	51	53.6 28.1	.43	24.96	1.15	1.09	-	49	1.3x10 ⁻⁴	-	Mixed flow
	62 01	48.7 1.80 5c.4 1.69	55.0	1.03	.782	-	-	6.06×10 ⁻⁵	186	Mixed flow	 	O F	.018	48 51	22.5	.40				<u> </u>	-		<u> </u>	Mand flow
0.040	95.5 71	54.1 2.16 54.3 1.57	ļ				_		-		319	0	010	45	26.8 19.2	.56	27.41	1.36	.768	ļ	48	1.2x10 ⁻⁴	<u> </u>	Mixed flow Popping based on O-graph
1,71	05.5	43.6 1.43	48.3	1.13	1.09		89	6.9x10 ⁻⁵	162	Mixed flow	320	jo		55.	18.0	25	18.5	1.40	.76		55	1.8x10 ⁻⁴	Ŀ	records
293	62 60	121 7.86 91.9 6.52	105.7	1.09	1.20	060	61	3.15×10 ⁻⁵	362	High speed movies recording, separated	321	F O	.010	58.5 56.5	19.3	.193 .255	18.6	1.40	,758	"	57	1.8×10 ⁻⁴	•	Popping based on O-graph records
294	60	54.3 1.57 46.4 1.67	49.8	1.24	.94	"	62	6.69x10 ⁻⁵	180	High speed movie recording, mixed	322	r O	.011	60.5 58.	19.4 18.38	.213	18.97	1.29	.818	.040	60	1.75×10 ⁻⁴	14.7	Mixed flow
295 F .034	53	72.5 2.16	67.2	1.44	.78		57	4.96×10 ⁻⁵	14.7	High speed movie recording popularies	323	F	.010	49.5	19.4	.194	19.10	1.46	.703	-	53	1.74×10 ⁻⁴		Popping based on O-graph record
296 F .033	54 57	60.7 2.03	54.7	1.15	1.09	.060	-	6.09x10 ^{~5}	14.7	High speed movie recording, popping observe	324	F	.009	51	17.9	.161	18.28	1.57	.623		50	1.82×10		Mixed flow
F .030	55	49.0 1.86 60.7 1.82	41.4	1.23	1.01	-	56	8.0x10 ⁻⁵	68	High speed movie recording popping observed	325	O F		49.5 71.5		.721	23.13	1.37	.782		69	2.2×10 ⁻⁵	-	Mixed flow
U .037	57 41	48.5 1.79 53.4 1.54				-	<u> </u>		<u> </u>	High speed movie recording	<u> </u>	O F	1	62.5 51	22.5	.923		-	ļ <u>.</u>	-	ļ	<u> </u>	 	Popping based on O-graph
O .035	51.5	45.2 1.61	49.2	1.20	0.96	.040	47	6.77x10 ⁻³	76	mixed flow	-	O	.039	57.5	21.6	.844	22.58	1.30	.847	├-	55	2.2x10 ⁻⁴	ļ	record
O .027	53	35.9 0.97	44.8	0.93	1.59	"	46	7.44×10 ⁻⁵	106	High speed movie recording mixed flow	 	0	.039	48.0	21.4	.827	22.27	1.30	.837		48		"	Popping based on O-graph record
1300	41 53	36.6 1.69 46.2 1.61	50.7	1.11	1.05		47	6.57×10 ⁻⁵	122	High speed movie recording mixed flow	328	10	.030	54.5 46.5		.715	22.70	1.31	.83	-	50	2.2x10 ⁻⁴	L"	Popping based on O-graph record
301	41 53	57.4 1.77 47.2 1.69	51.6	1,16	1.05		47	5.46×10 ⁻⁵	100	High speed movie recording mixed flow	329	FO	10-2-	52.5 46.5		.704	22.2	1.27	.90		49	2,25×10	•	Popping based on O-graph record
302 F .031	43	57.4 1.77	46.5	1.07	1.74	-	49	7.17x10 ⁻⁵	192	High speed movie recording mixed flow	330	F	.030	52.5	23.8	.705	22.2	1,27	.90		50	2.25×10	٠.	Popping based on O-graph record
F .023	55 37	35.5 1.02 42.4 .98	35.5	1.04	1.41	-		9.4×10 ⁻⁵	14.7	High speed movie recording	331	O F	.039	47.5 60.5	30.7	1.20	28.4	1.24	.92		62	1.75×10	-	Popping based on O-graph
F .01	37 42	28.9 .69 16.2 .16						2.3×10 ⁻⁴	-	popping observed High speed movie recording		O	1	64.0		2.38		-	 	-	-	<u> </u>	+	record Popping based on O-graph
305 O .01	42.5	12.8 .13	14.5	1.0	1.27	<u> </u>	42			mixed flow	<u> </u>	0	.061	63.0	33.8	2.06	38.3	1.11	1.16	-	63	1.3×10 ⁻⁴	75	record
0 .01	38.5	16,2 .16 16,1 .20	16.14	1.26	0.80	"	41	2.06×10 ⁻⁴	"_	High speed movie recording popping observed		+-	.071	63.0		2.81	44.4	1.13	1.12	<u>"</u>	63	1',1x10 ⁻⁴	80	Popping based on O-graph record
307 F .026	32 32	48.5 1.26 52.3 2.15	50.8	1.58	0.59	"	32	6.5xl0 ⁻⁵	"	High speed movie recording mixed flow	334		.073	63.0	57.4 33.5	4.20 2.01	46.6	.822	2.08	н	62	1.0x10 ⁻⁴	95	Popping based on O-graph record
																					-			·

Test W T V M	Ī	0/F	M _I /M	c D	Ī,	D/V	P	Comments	Te	st w	Т	l v	М	v	0/F	M _f /M _c		= 1	D/V .	P	Comments
No. F 039 60.5 30.7 4.98	a		1.57	+		- 4	Ι.	Popping based on O-graph	_ No	1 .064	+	<u> </u>	3.24	a_	-			Ta	1.08×10		Popping based on motion
O 048 64.0 26.4 1.26	37.2		1.37	1000		1,3x10	93.	record	361	U .065		41.9	2,72	46.2	1.02	1.19	.060	55	1.000010	163	picture & O-graph recording
336 F .055 64.0 43.6 2.39	38.5	1.11	1.15	i.	64	1.2x10 ⁻⁴	70		362	F .065			3.35	45.6	1.12	1.14		54	1.09x10	105	
O .061 63.0 34.0 2.08	<u>.</u>	,	-				-		ļ	0 .973		40.3	2.94		ļ		 	-			
337 0.065 61 36.2 2.35	47.4	. 89	1.78		56	1.0x10 ⁻⁴	,80	. "	363	F .064		50.5	3.23	46.6	1.23	0.93	•	52	1.07×10	١.	•
338 F .063 89 49.7 3.13	45.9	1.24	.937		¹ 87	1.0×10 ⁻⁴	85		364	F .065			3.35	47.5	1.23	0.95	ļ. —	61	1.05×10		,
O .078 86 43.0 3.34		·	1	<u> </u>			,		304	O .080	\$9.5	44.2	3.54	47.5	1,23	0.33					
339 F .074 96.5 57.8 4.28 O .082 108 45.0 3.69	51.1	,1.11	1.16	, "	100	9.7x10 ⁻⁵	125	h h	365	Г.064		_	3.23	46.5	1.22	0.96		61	1.07x10	.	н
340 F .063 117 49.B 3.14	42.0	ii ne	1.31	1	1112	4	<u> </u>		 	D .078	$\overline{}$		3.38		· ·						
O .066 108 36.3 2.40	42.9	11.05	1.31	1	1112	1.2×10 ⁻⁴	75.		366	0 .081	59.5	$\overline{}$	3.64	47.9	1.25	0.92	"	61	1.0x10 ⁻⁴	"	·
341 F 063 107 50.1 3.16	45.1	1.16	1.06		103	1.1x10 ⁻⁴	75		367	₽ .065	79.5	51.6	3.35	47.7	1.25	0.93	"	77	1.04x10-		
341 O 073 99.5 40.7 2.97		+-		-	1				<u> </u>	0 .081	76.0		3,62					1			
342 O 075 99 41.9 3.18	45.6	1.21	.994		102	1.0x10 ⁻⁴	145	.	368	F .062	57.5	49.1	3.04	46.5	1.29	0.85		69	1.07x10	165	
343 F .063 118 49.8 3.14	44.8	1.16	1.06	-	110	1.1×10 ⁻⁴	1145	,	369	F .063	$\overline{}$		3.14	45.0		0.93	<u> </u>	67	1.08×10 ⁻⁴		Popping based on O-graph
O .073 103 40.5 2.96	11.0	1	1.00	<u> </u>		1	.43		369	0 .078	69.5	43.3	3.38	46.2	1.24	0.93		6′	1.00x10		recording
344 F .063 81 49.8 3.14 O .078 79.5 43.3 3.89	49.9	1.24	.807		80	1.0×10 ⁻⁴	145	"	370	F .062	64.0		3.04	44.8	1.21	0.98		64	1.12x10	,	
7 002 01 40 1 2 04			_	+-			-	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		0 .075			3.10					\vdash			
345 O .078 77.5 43.3 3.38	45.9	1.26	.899	1	79	1.0×10-4	165		371	F .063			3.11	43.1	1.06	1.23	"	63	1.16×10-	"	•
346 F .063 75.5 49.8 3.14	46.2	1.28	.88		76	1.08×10 ⁻⁴	165		372	F .062	$\overline{}$		3.01	41.3	0.98	1,45		55	1.21×10		-
0.080 75 44.5 3.56				<u></u>					3/2	0 ,061	54.5	34.0	2.07	415	0,30	1175		33		<u>.</u>	
347 F .063 79.5 48.8 3.14 O .079 78 43.54 3.42	46.5	1.25	.918	"	79	1.08x10-4	165	•	373	F .062	_	59.8	3.71	53.0	1.23	1.03 .	055	63	8.64x10-5	95	Popping based on motion
348 F 063 83.3 49.8 3.14	41.3	911	1.72	060	83	1.21x10 ⁻⁴	165	Popping based on motion	374	D .076	64	52.1	3.60	54.8	1.19	0.99	.055	63	8.36x10 ⁻⁵	95	Popping based on motion
O 057 82.5 31.8 1.83		7		1	-			picture & O-graph recording	3/4	U .074	62.5	50.6	3.74	34.0	1.13	0.33	.033	03	0.30210	"	picture & O'graph recording
349 F 1064 91.0 50.5 3.23 O 060 89.5 33.0 1.97	42.1	.931	1.64	-	91	1.19x10 ⁻⁴		-	375	F .062	64	59.8	3.71	57.7	1.27	0.79	•	63	7.94x10 ⁻⁵	•	·
F 054 01 0 50 5 2 22			 	ļ.,	-	1.12x10 ⁻⁴			-	O .079	62.5	54.2	4.72				-	-			
O 072 93.5 39.7 2.85	44.8	1.12	1.13		92	1.12x10	165	"	376	0 080	62.5	54.4	3.71 4.35	56.8	1.29	0.85	•	63	8.06x10 ⁻⁵	165	•
351 F 063 72.0 49.8 3.14	48.3	1.35	7.82		71	1.03×10 ⁻⁴		"	377	F .062	81.5	60.1	3.73	56.5	1,27	0.88		72	8.1x10 ⁻⁵	175	
O 085 69.5 47.2 4.02 F 062 71.5 49.1 3.04					_		_		ļ	0 .079	64.5	53.6	4.23			****					
352 C 075 67.5 49.1 3.04	44.8	1.21	.98	"	69	1.1x10 ⁻⁴	165	Popping not observed	378	Γ .062 O .081	89 58	55.5	3,75 4.50	57.7	1.31	0.83	•	77	7.9x10 ⁻⁵	•	•
353 F .062 70 48.6 3.02	45.2	1.23	.938	-	68	1.1x10 ⁻⁴		Popping based on motion	379	F .062	B1,5	60.5	3.75	57.3		0.86	_	94	7.99×10 ⁻⁵		
0 076 66 42.4 3.22		11.23	.550	\vdash	-			picture & O-graph recording	3/9	0.080	104	54.7	4.38	57.3	1.29	0.86	_	34	7.99X1U		
354 F .064 49 50.5 3.23 O .076 50 41.9 3.18	45.8	1.19	1.02	-	50	1.1×10 ⁻⁴	٠.	-	380	F .063	105	60.9	3.84	56.7	1.24	0.93	-	110	8.0x10 ⁻⁵	•	.
355 F .064 47 50.5 3.23		 						No seeding to the seeding to		O .078	107	60.9	3.84				_		5		
0.079 46.5 44.0 3.48	46.9	1.23	0.93	-	47	1.06×10 ⁻⁴	175	No popping was observed	381	0 .080	114	54.4	4.35	57.3	1.27	0.88		111	7.99x10 ⁻⁵		
356 F .064 49 S0.5 3.23	43.3	1.03	1.35		50	1.15x10 ⁻⁴	165	Popping based on motion Picture & O-graph recording	382	F .063	110	60.9	3.85	53.2	1.21	1.08		111	8.6x10 ⁻⁵	$\overline{\cdot}$	ж
O .066 50 36.3 2.40 F .062 51 48.9 3.03		-						, locale a O-graph recording	ļ	0 .076	_	52.1	3.56				\vdash				
357 0.075 50 41.4 3.11	44.8	1,21	0.97	"	51	1.11x10 ⁻⁴	"	"	383	F .062	66	55.9	3.73 4.58	57.7	1.32	0.81		65	7.94×10 ⁻⁵	-	• 1
F .064 51 50.5 3.23	45.8	1.19	1.02		12	1.96×10 ⁻³			384	F .062	77.5	60.4	3.74				-				
358 O .076 50 41.9 3.18			L <u>.</u>	_					307	0 .080	87.5	54.7	3.78	52.9	1.29	0.99		83	8.66×10 ⁻⁵		·
359 F .064 47 50.5 3.23 O .076 48 41.9 3.18	45.8	1.19	1.02	"	48	1.96×10 ⁻³	-		385	F .062	77.5	60.4	3.74	56.2	1.26	0.91	•	77	8.15×10 ⁻⁵		•
E 026 52 5 50 4 4 50	40.2	070	1 02		-	1 02=10=4				0 .078	76	52.9	4.13		لــــا					Ш	
0.060 56.5 33.0 1.98	48.3	.079	1.02		54	1.03×10-4	-	"	i												

						16			-																
Test w	T	V	М	v	O/F	M _f /M _o	D	Ta	D/V _a	Р	Comments	To No		w	Г	V	M	ν̄ _a	O/F	Mf/Mo	D	T _a	D∕√a	P	Comments
388 F .058	51.0 47.5	45.9 41.8	2.66	43.5	1.29	.848	055	49	5.!x10 ⁻⁵	175	Popping based on O-graph recording	414	r	.069	105	54.3 38.6	3.74	46.6	1.01	1.39	"	100	1.07x10 ⁻⁴	80	Popping based on motion picture and O-graph
389 F 088	51.0 47.5	45.9 36.5	2,66	40.88	1.13	i.10		62	5.5×10 ⁻⁵	175	,	415	0	.069	108	54.3	3.74	48.7	1.14	1.09	-	103	1.03x10 ⁻⁴	85	Popping based on motion picture and O-graph
390 F .058	51.0 47.5	45.9 55.2	2.66	51.65	1.71	.484		49	4.3×10 ⁻⁵	175	"	416	F	.068	110	53.5	3.62	47.9	1.13	1.11		105	1.04×10 ⁻⁴	80	Popping based on motion picture and O-graph
391 F .075	51.5	59.7	4.51	53.9	1.17	1.04		49	4.1x10 ⁻⁵	195		417	1	.077	116	42.6 55.1	3.27	47.8	1.05	1.13	-	113	1.04x10 ⁻⁴	80	Popping based on motion picture and O-graph
392 F .076	51.5	59.7	4.51	53.8	1.16	1.06	-	49	4.2×10 ⁻⁵	195		418	O T	.073	109	40.7 55.1	3.85	47.1	.95	1.43		109	1.06x10 ⁻⁴	80	Popping based on motion picture and O-graph
F .075	60.0	59.1	4,42	4B.5	.891	1.84		59	4.6×10 ⁻⁵	415	Separated flow based on motion recording	-	O F	.069	102	38.6 54.3	3.74	45.8	.98	1.49	-	hna	1.09×10 ⁻⁴	80	Popping based on motion
394 F .090	62	36.5 71.0	6.37	65,1	1,21	.967	.060	59	7.67×10 ⁻⁵	235	Mixed flow	419	O F	.067	102	37.2 54.3	2.49 3.74	45.1	.91	1,70	-	105	1.11x10 ⁻⁴	_	picture and O-graph Popping based on motion picture, Pc not available
395 F .090	56 52.5	70.7	6.59	50.6	1.02	1.37		49	8.25×10 ⁻⁹	225	Mixed flow	420	0	.063	101	34.9 54.3	2.19					-			Popping based on motion
O .091	45 45	50.7 71.1	6.4		-		-	-				421	0			33.0 54.3	1.97	44.6	.87	1.90	_	117	1.12×10 ⁻⁴		Popping based on motion
O .108	37.5	59.8	6.5	65.0	1.20	1.52		46 76	7.69×10 ⁻⁵	235	"	422	0	.057		31.8	1.83	64.1	.63	2.04	Ë	111	1.13×10 ⁻⁴		picture, Pc not available
O .107	62.5	59.4 71.0	6,37		-			_		245		423	0		109	51.7	4.82	52.0	1.35	.78		110	9.46×10 ⁻⁵	105	Popping based on motion picture and O-graph
398 O 121	71 75.5	56.3	6.84	62.6	1.35	.931	"	71	7.98×10	235	"	424		.069	103	54.3	3.74 4.56	52.2	1.32	.82	"	107.	1.10x10 ⁻⁴	95	Popping based on motion picture and O-graph
399 F .090 O .098 400 F .090	79.5	54.2 71.1	5.30 5.41	62.1	1.09	1.19		77	8.04×10 ⁻⁵	235	"	425	r o	.067		53.5 37.2	3,62	45.4	.99	1.45		82	1.11×10 ⁻⁴	95	Popping based on motion picture and O-graph
U .096		53.0	5.07	61.8	1.06	1.24	060	81	8,0x10 ⁻³	235	Mixed flow	426	F O	1144	64.5	53.5 36.2	3,62 2,39	44.9	.97	1.15	.060	73.	1.10×10 ⁻⁴	95	Popping based on O-graphs
401 F .090	87 82.5	71.1 51.2	6,41 4,72	61.0	1.02	1.36	"	85	8.19x10 ⁻⁵	235		427	7	.069	81.5	54.7 34.9	3.7¢	45.3	.9!	1.72	"	80.4	1.10x10 ⁻⁴		Popping based on O-graphs
402 F .087	98.5	51.2	6.24	61,1	1.60	1.32		90	8.18×10 ⁻⁵	235		428	F	.069	_	54.7	3.78	47.1	1.04	1.33	"	106	1.06×10 ⁻⁴	-	Popping based on O-graphs
403 F .077	97.5	61.0	4.72	54.7	1.16	1.05	-	100	9.13×10 ⁻⁵	255	u	429	.,	.069	114	54.7	3.78	45.7	.95	1.59	ļ.,	110	1.09x10 ⁻⁴	41	Popping based on O-graph and motion pictures.
404 F .076	108 96.5	50.3	3.85	49.8	1.16	.882		102	1.0x10 ⁻⁴	215	ıı .	430	F	.060		36.3 58.2	3.48	53	1.19	1.0	.055	71.	8.64x10 ⁻⁵	185	Popping based on O-graph and motion picture
405 F .087	112.	68.8 48.0	5.93	58.4	.993	1.44		106	9.9×10 ⁻⁵	235		431	О Г	.060	$\overline{}$	48.7 58.5	3.48	52.2	1.14	1.1	- "	₩		175	Popping based on O-graph
406 F .083	66	66.1 56.0	5.49	60.6	1.02	.98		64	8.24×10 ⁻⁵		п	432	O F	_	_	74.0	3.20 5.64			<u> </u> 	\vdash	-	 	-	No popping. (Mixed)
407 F .083	66 62.5	65.4	5.42	58.5	1.14	1.09		64	8.54×10 ⁻⁵		o .	432	0 F	1.030	92.5 106	65.4 73.5	5.55	69.4	1.26	.89	ļ.,	_	6.69×10 ⁻⁵	215	No Popping (Mixed)
408 F .088	66	65.4	5.74	61.2	1.11	1.08		66	8.17x10 ⁻⁵		п	434	O F	.094	97.5 114	64.4 64.5	6.03	57.0	1.20	1.06		 	8.03×10 ⁻⁵	176	No constant (Marcoll)
O .098	62.5	54.2 65.4	5.42	57.3	1.09	1.21	-	64	8.73×10 ⁻⁵	-	n	434	0	+-0.2	101	51.0 60.4	3.79	···			 	<u> </u>	ļ <u> </u>		No popping (Mixed) Popping based on
O .089	62.5 85	49.8	3.74	45.4	.946	1.62	<i>-</i>	81	1.10x10 ⁻⁴	185	Popping based on O-graph		O	.073	96	50.1	3.68	54.8	1.18	1.02	Ë		8.36×10 ⁻⁵	/3	O'graph recording Popping based on
O .065	94.5	36.l 55.l	3.85	45.2	.877		-	92	1.11x10 ⁻⁴	175	& motion picture recording	436	0	.066	72.5	45.2	3.68	52.3	1.08	1.23	<u> </u> "		4.3×10 ⁻⁵	95	O-graph recording
O .061	97	33.9 55.1	2,08									437	0	.066	68.0	45.2	3.0	52.3	1.08	1.23		73	4.3×10 ⁻⁵		Popping based on O-graph recording
412 O .064 413 F .070	86.5 89	35.6 55.1	2.28	45.7	.919	1.69	060	92	1.09x10 ⁻⁴	190	Panning based as O see the	438	0		99	50.8	2.27	52.8	1.52	.62	.040	101	6.3×10 ⁻⁵	45	Mixed flow
U .072	82.5	39.9	2.87	47.13			.000	03.3	XIU	30	Popping based on O-graph	4 51	, ,	.02	$\overline{}$	53.4 49.5	1.55	51.3	1.32	. 81	.04	102	6.5 x 10 ⁻⁵	45	Mixed flow

Test	Tv	v T	Т	v	м	⊽_	O/F	M _f /M _o	D	Ťa	D/V _a	P	Comments	Tes	t	w	r	v	м	v,	O/F	M _f /M _o	D	Ťa	D/V _a	P	Comments
440 F		33 102 145 97.		59	1.44	,3,3	1.3	.76		80	'./xld"'	i	Mixi d tlow	46b	_	01:3 09:9	52.5	65.7	5.45	59.8	1.19	1.00	-	49	8.36×10 ⁻⁵	435	Separated flow based on motion picture recording
441 F	.0	130 97 138 91	+	53.4	1.55	50.8	1.30	. 34	-	95	6.5×10 ⁻⁷	-	Mixed flow	467	F,	063	52.5	65.7	5.45	60.27	1.21	.97	-	49	8.3×10 ⁻⁵	455	Separated flow based on motion picture recording
442 F	.0	29 116	7	53.4 58.5	1.55	\$6.5	1.""	.5н		111	5.9×10 ⁻⁵	-	Mixed flaw	468	F.	101 083	110	65.7	5.63 5.45 6.93	63.5	1.35	. 79	•	109	7.87×10 ⁻⁵	"	Separated flow based on motion picture recording
443 <u>F</u>	.0	29 125 42 114	1	53.4 54.12	1.55	\$3.5	1.45	.68		1111	6.2×10	-	Separated flow due to OX boiling.	469	1	112 08/ 109	116	68.8	6.0	65.3	1.25	.91		115	7.78×10 ⁻⁵	"	Separated flow based on motion picture recording
444 F	.0	29 108	Ť	53.4	1.55	57.4	1.60	.55	-	108	5.8×10 ⁻ '		Mixed flow	470		087 105	119	68.6 58.2	6.0	63.1	1,21	98	4	115	7.93×10 ⁻⁵	445	Separated flow based on motion picture recording
445 F	.0	31 122 39 114	7	57.4	1.80	53.3	1.04	.93	-	117	6.2x10 ⁻⁵		Separated flow due to OX boiling	471	F.	067	92	52.8 46.6	3.53	49.3	1,26	.90	"	68	1.01×10 ⁻⁴	100	Popping based on motion pictures & O-graph record
446 F	ف ا	31 126 38 116	7	57.4 49.3	1.79	52.9	1.24	.96		121	6.29x10 ⁻⁵		Separated flow due to OX boiling	472	F.	067	72	52.6	3.50	47.7	1.15	1.07	-	64	1.05×10 ⁻⁵		Popping based on motion plotures & O-graph record
447 F	.0	35 122 39 108	\dashv	66.6 50.6	2,42 1,98	58.3	1.08	1.22		114	5.7×10 ⁻⁵	.,	Mixed flow	473	Γ,	067	70 63	52.6	3.50	49.2	1.26	.90		66	1.02×10 ⁻⁵		Popping based on motion pictures & O-graph record
448 F	.0	36 114 42 104	1	66.E 54.1	2.42	59.9	1.16	1.06		110	5.56×10 ⁻⁵	55	Mixed flow	474	F.	067	62 59.5	52.6	3.50	47.3	1.16	1.06		61	1.05×10 ⁻⁴		Popping based on motion pictures & O-graph record
449 F	.0	36 116 42 106	7	66.7 57.2	2.42	59.9	1.16	1.06	-	110	5.55x10 ⁻⁵		Mixed flow	475	F.	042	64	77.9	3.30	68.3	1.09	1.20	040	66	4.08×10 ⁻⁵	630	Separated flow based on motion picture recording
450 F	.0	36 118	7	56.6 54.2	2.47	59.9	1.16	1.06		110	5.56x10 ⁻⁵		Mixed flow	476	F	045 057	_64 68	81.7	3.64	77.09	1.28	.87		66	4.32×10 ⁻⁵	80	Mixed flow based on motion picture recording
451 F	.0	37 106 35 101	7	68.2 45.7	2.53	57.2	1.13	2.57	7	103	5.8×10~5	45	Mixed flow	477	F	045 048	66	83.2	3.77 2.91	72	1.05	1.29	· ·	64	4.6×10~5	85	Mixed flow based on notion picture recording
452 F	1.0	35 108 34 97.		65.1 44.4	2.3	54.8	.97	1.51	.040	103	6.08×10 ⁻⁵	45	Mixed flow	478	F .	038	66 59.5	70.	2.67	67.4	1.33	.807	.040	62	4.94×10 ⁻⁵	640	Mixed flow based on motion picture recording
453 F	+	35 105 35 91	7	65.1 44.9	2,30	55.1	.98	1.47		98	6.05×10 ⁻⁵	50	Mixed flow	479	$\overline{}$	067	71.5	52.8	3.53	49.3	1.25	.905	.060	69	1.01×10 ⁻⁴	100	Popping based on motion pictures & O'graph record
454 F	-	39 132 39 114	_	40.9 49.8	1.92	60.4	1.0	1.43	-	123	5.52×10 ⁻⁵	65	Mixed flow	480		068	6 4	53.7 48.4	3.64 4.22	50.7	1.28	.86		66	9.86×10 ⁻⁵		Popping based on motion pictures & O-graph record
455 F	_	37 134 53 122	\neg	68.2 67.7	3.55	67.9	1.41	.712	-	127	4.9x10 ⁻⁵	50	Separated flow due to OX boiling	481		045	132 128	35.1 26.5	1.56	30.8	1.06	1.24	"	129	1.62×10 ⁻⁴	23	Popping based on motion pictures & O-graph record
456 F	+	36 137 45 129	-	66.6 58.5	2.42	62.1	1.25	.913		133	5.36x10 ⁻⁵	55	Separated flow due to OX boiling	482		044	141	34.9 25.4	1.54	30.1	1.03	1.33		138	1.66×10 ⁻⁴	147	Popping based on O-graph recording
457 F	→••	36 144 47 135	\neg	66.3 60.5	2.42	63,1	1.29	.852	".	139	5.27×10 ⁻⁵		Separated flow due to OX boiling	483		035	60.5 51.5		2.30	54.3	.946	1.60	l"	66	6.14×10 ⁻⁵	175	N ₂ H ₄ +5%NH ₄ NO ₃ . Popping based on O-graph record
458 F		35 144 47 131		64.1 60.0	2.24	61.7	1.33	.801	"	137	5.4x10 ⁻⁵	·	Separated flow due to OX boiling	484	-	035	50.5		2.24	58.6	1.20	.986	"	56	5.68×10 ⁻⁵	185	11
459 F		36 147 45 137	$\overline{}$	62.3 58.5	2.11	60.1	1.34	.797	"	142	5.55×10 ⁻⁵	-	Separated flow due to OX boiling	485		034	60.5 57.5	_	2.13	57.9	1.23	.938	•	59.	5.75×10 ⁻⁵	220	N ₂ H ₄ +5%N ₂ H ₄ NO ₃ Rough chamber pressure
460 F	-	84 64 68 61.	_	66.3 37.9	5.56 2.59	53.6	.814	2.15	D 60	63	9.3x10 ⁻⁵	470	Separated flow based on motion picture recording	486		017	58.5 54.5	_	1.07	56.9	2.51	.471	.040	55	5.86x10 ⁻⁵	90	N ₂ H ₄ +5%NH ₄ NO ₃ Mixed flow
461 F	+	_	_	66.9 65.1	5.68 7.19	64.7	1.34	.789	"	64	7.7x10 ⁻⁵	565	Separated flow based on motion picture recording	487		034	58.5 56	63.1 54.1	2.16	58.1	1.22	.952		57	5.73×10 ⁻⁵	100	
462 F	+	87 66. 87 62.		68.6 48.4	5.99	5e.7	1.00	1.42		64	8.5×10 ⁻⁵	535	Separated flow based on motion picture recording	488		035 042	58.5 56	63.8 54.1	2,22	58.5	1.21	.970	•	57	5.69×10 ⁻⁵	100	
463 F	+			68.8 46.6	5.99 3.91	57.9	.96	1.54	"	65	8.6×10 ⁻⁵		Separated flow based on motion picture recording	489	Г О	035 042	58.5 56	63.8 54.1	2,22	58.5	1.21	.970	"	57	5.69x10 ⁻⁵	100	•
464 F	~	87 69. 02 66.		68.2 56.5	5.99	62.2	1.17	1.04		67	8.0×10 ⁻⁵	\$85	Separated flow based on motion picture recording	490	F	024	62.5 62.5	43.4 39.8	1,02	41.0	1.31	.83		63	8.12×10 ⁻⁵	150	
465 F		87 64 10 67	1	68.8 61.3	5,99 6,74	64.6	1.26	.89	.060	66	7.7x10 ⁻⁵	475	Separated flow based on motion picture recording	411	F O	. 024 . 02:1	56.5 51.5	38.0	1,02	49.4	1.25	.91	.040	54	8.24×10 ⁻⁵	30	N ₂ H ₄ +5% NH ₄ NO ₃ Mixed flow

	_											т	1			
Test No.		w	т	v	м	v _a	O/F	M _f /M _o	D	Ťa	D/V _a	P	Comments			
492	F	.023	60	41.4	١٤،٠	ذ. 40.	1,37	. /5		57	·*-01x65.					
132	0	.031	14. 5	3: .1-	1.23	,,,,,			_			_				
493	F	.022	60	40.8	.907	36.7	1.16	1,06		5.0	J.07x10 ⁻³	-	N2H4+5%NH4NO3			
493	0	.026	56	33.2	. 85.2								Mixed fl.iw			
494	F	.022	_62_	40.٤	.907	37.1	1.16	1.04		59	×10-5	. 1				
	이	.026	56	35.9	.£71											
495	F			43.4	1.02	40.5	1.20	. 90	.]	60	8.22×10 ⁻⁵	35				
	0	.029	.56	32.3	1.14		ļ					-				
496	F	.024	_66	43.4	1.02	37.8	1,21	. 38	-	61	1:.38x10 ⁻⁵					
	0	.029	56	36.a	1.05		<u>└</u>	_				_				
497	F	.024 66 44.8		44.8	1.09	42.1	1.27	.83	-	60	7.92×10	30	N ₂ H ₄ +5%UDMH			
	0	.031	56	39.6	1.23								Mixed flow			
498	P	.024	66	44.6	1.09	42.1	1.27	. 20		61	7.3×10 ⁻⁵					
	0	.031	5.8	39.8	1,23							_				
433	F	.024	64.5	43.4	1.02	41.4	1.31		- 1	20	8.05×10 ⁻³	- 1				
	0	.031	56	39.8	1.23				Ш			$oxed{oxed}$				
500	F	.024	66	43.4	1,02	41.4	1 1.31	.83	-	60	8.06x10 ⁻⁵	۱.				
	0	.031	56	39.8	1.23				\sqcup							
501	F	.021	74	38.3	. 796	34.2 1.1	1.14	4 1.09		67	9.7×10 ⁻⁵					
	0	.024	62.5	30.5	.726			-			<u> </u>	<u> </u>				
502	F	.020	69.5	37.3	.756	34.9	1.27	38.		66	9.5×10 ⁻⁵					
302	0	.026	64.5	33.2	.853		<u> </u>					<u> </u>				
503	F	.021	74	38.3	.796	35.5	1.23	.93	۱.,	69	9.4x10-5	١	,			
303	6	.026	66	33.2	.853	3.7.0		1			l	<u>i,</u>	<u></u>			
504	F	,017		31.2	.526	32.7	1,54	.598	040	69	1.02×10 ⁻⁵	30	N2H4+S%UDMH			
304	0	.026		33.7	.870	<u> </u>	—	—	┼	╄-		1	Mixed flow			
505	F	.009	_	17.23	.162	19.47	1.77	.488	.040	a2	1.71×10 ⁻⁴	14.7	N ₂ H ₄ +5%UDMH-Separated flow due to OX boiling			
	10	.016	72	20.68	.331	19.47	1.77	.400	.040	00			How due to ox borning			
506	F	.012	_	21.7	,256	21.52	1.4	.721	.040	72	1.55×10-4	"				
	ᅝ	.017	72	21.4	355		-	-	-			+	N H 459 MMH Cor			
507	F	.007	71	27.41	.136	37.42	2.2	.296	.027	71	6.03×10 ⁻⁵		N2H4+5% MMH, Separated flow due to OX boiling			
	0	.015		41,37	.629	1	+	-	₩	-		┰				
508	F	.010	70	39.2	.380	39.9	1.4	.65	000		5.6 ×10 ⁻⁵		N ₂ H ₄ +5% MMH Mixed flow			
	10	.014	68	40.34	-581	1			.027	69		+	Mixed How .			
509	E	.013	68	53.23	.703	49.8	1.2	7 .89	.027	65	4.5×10 ⁻⁵	-				
	0	1.017	017 68 47.07 .79		.791	1,5,5	1.,-					_				